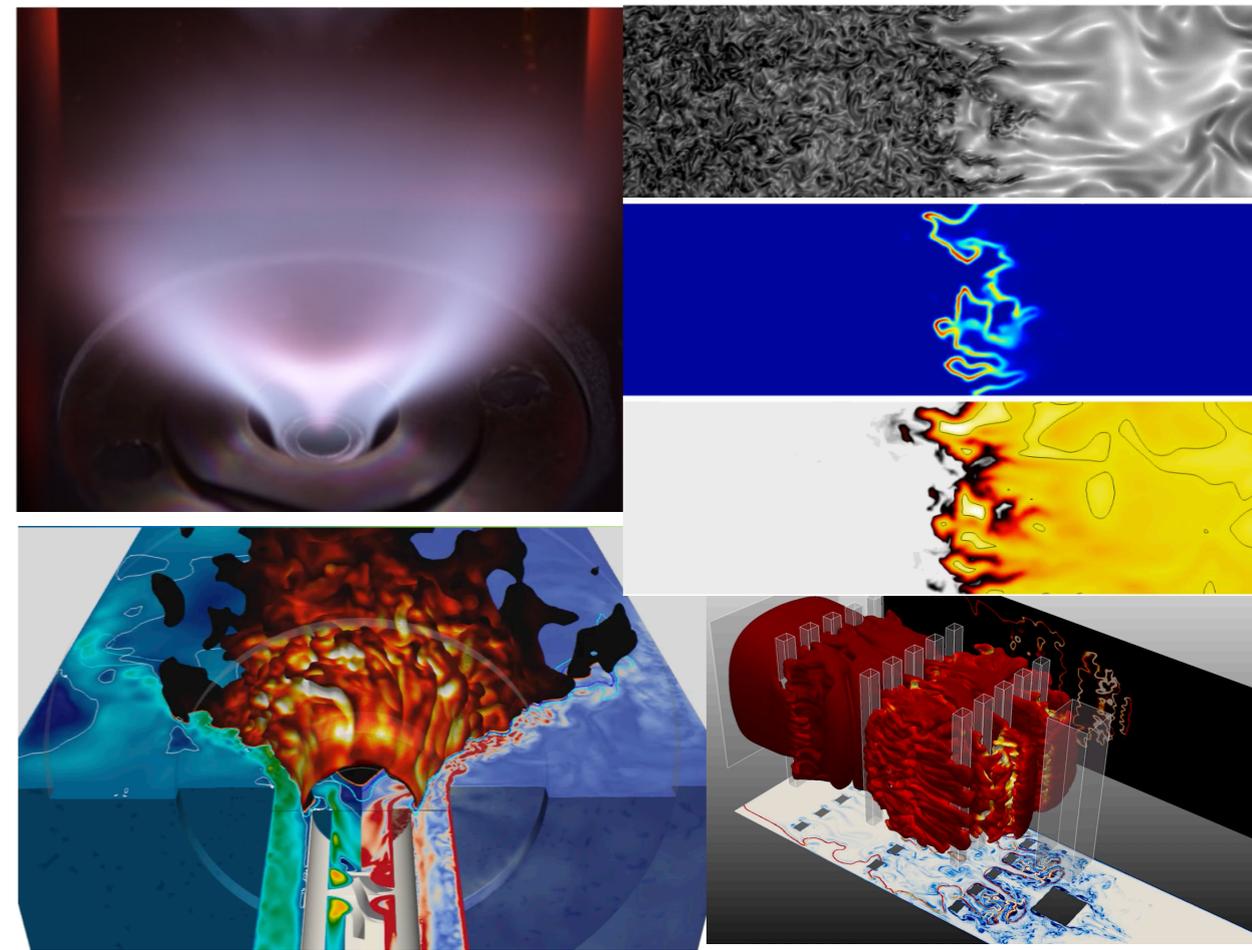


CFD AND HYDROGEN COMBUSTION

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- Stanford University
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C. Brunet, S. Richard **Safran Helicopter Engines**

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CAN CFD DO WHAT WE EXPECT FOR H2 ?

EVERYONE PLANS TO USE CFD TO EXPLORE ALL POSSIBLE ENGINE GEOMETRIES AND FUEL CHARACTERISTICS BEFORE ENGINES ARE BUILT. SAME EXPECTATION FOR SAFETY SCENARIOS

THIS IS DIFFERENT FROM EXISTING KEROSENE-AIR ENGINES WHERE CFD EXPERTS NOW HAVE PLENTY OF EXPERIENCE BASED ON MEASUREMENTS, TRIAL, ERROR AND ... TUNING.

IS H2 EASIER ?

ARE OUR CODES READY FOR THIS ?...

CFD OF ENGINES HAS MADE HUGE PROGRESS

First full engine computation with large-eddy simulation

Project FULLEST - C. Pérez Arroyo et al. - 2020

This is an engine burning kerosene.
What happens if we replace it by H₂ ?



Prix ATOS Joseph Fourier 2021

OUTLINE OF PRESENTATION

- ★ **Thermodiffusive instabilities and first models in LES
(2022 Center for Turbulence Stanford Summer Program +
J. Gaucherand, V. Coulon, A. Aniello PhDs at CERFACS
and IMFT)**
- ★ **Walls and hydrogen chemistry (L. De Nardi PhD)**
- ★ **Flame stabilization processes at the lips of injectors for
hydrogen flames (J. Bertsch, M. Chen, A. Aniello PhDs)**

FROM A CFD POINT OF VIEW

★ The good news for CFD experts is that the chemical schemes for H₂ air flames are:

→ well known

→ simple

★ Unfortunately, with H₂, we have other... issues

IS H2 SPECIAL FOR CFD ?

COMPUTING H2 SYSTEMS PUSHES US IN TWO DIRECTIONS CFD DOES NOT LIKE:

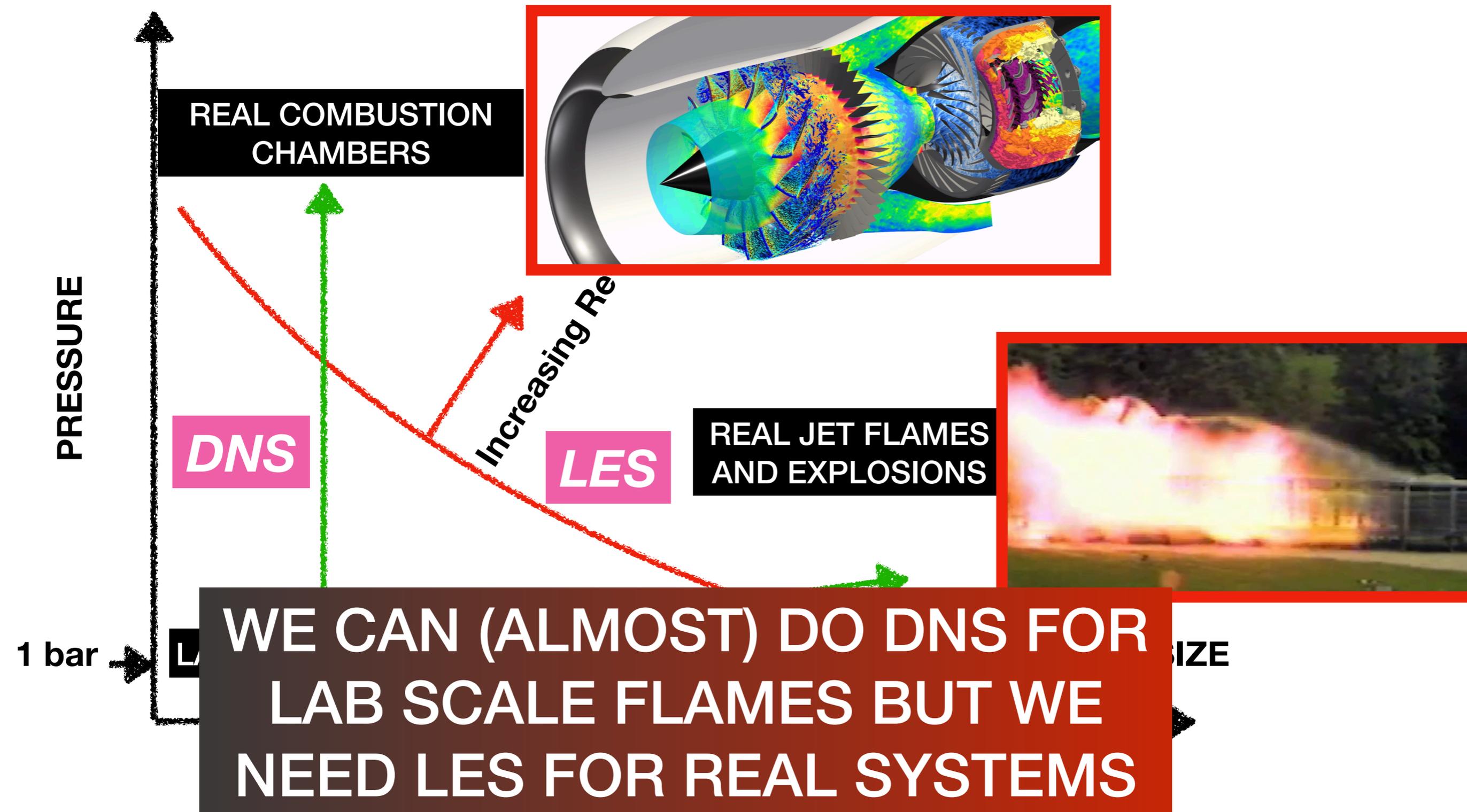
- Small chemical times and thicknesses***
- Higher pressure (for engines)***
- Larger domains (for explosions)***

The two tough axis for H2 air flames:

Reynolds = SIZE * SPEED / nu

where nu = nu0 Po/P

A Re=ct line in a P-SIZE diagram is a parabola



BUT THERE ARE OTHER ISSUES:

- *H2 flames are often a mix of premixed and diffusion flames: not that many models can do 'multimodes'***
- *H2 diffuses very rapidly, creates specific instabilities such as thermodiffusive cells: one paper submitted to Comb. Flame per week -> we love this one***
- *H2 chemistry at walls has problems...***
- *The stabilization of H2 flames at lips is not understood yet***

THERMODIFFUSIVE INSTABILITIES

H₂ is a small molecule: it diffuses faster than heat.

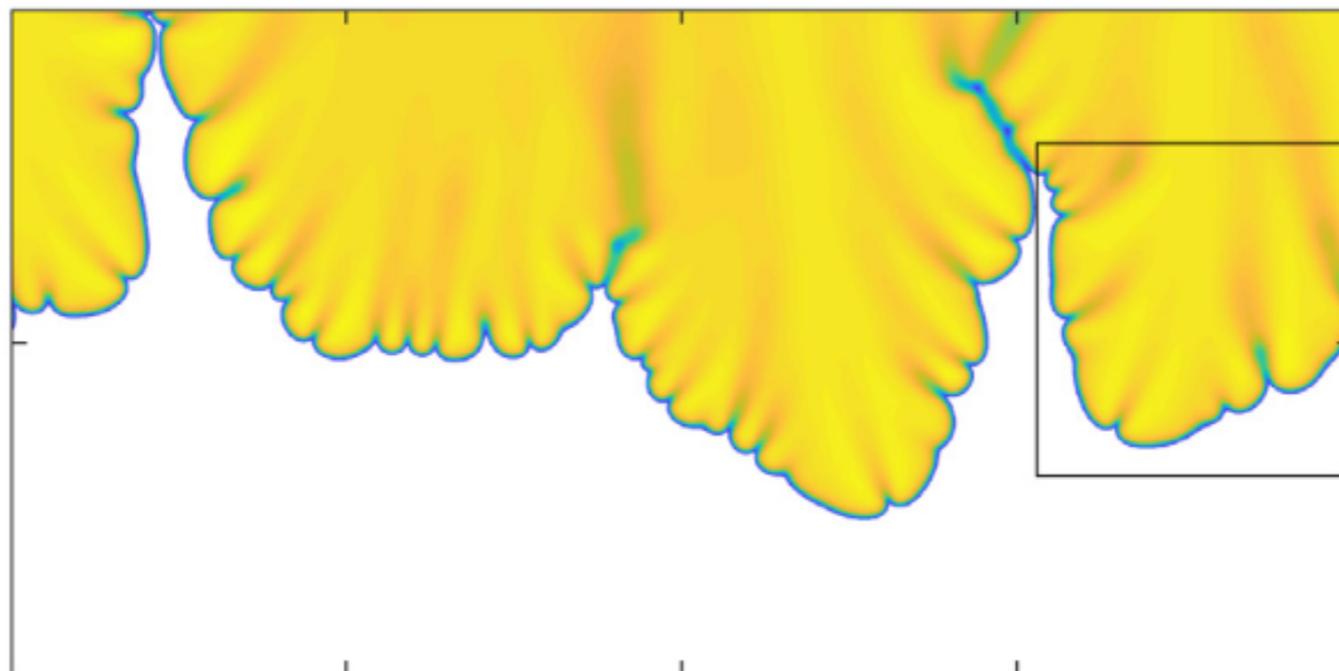
This is quantified through the Lewis number:

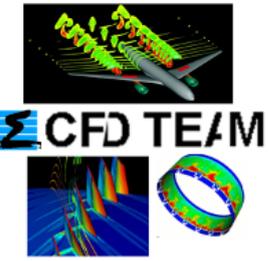
$$L_e^k = \lambda / (\rho C_p D_k) = D_{th} / D_k$$

For H₂: $L_e^k = 0.3$

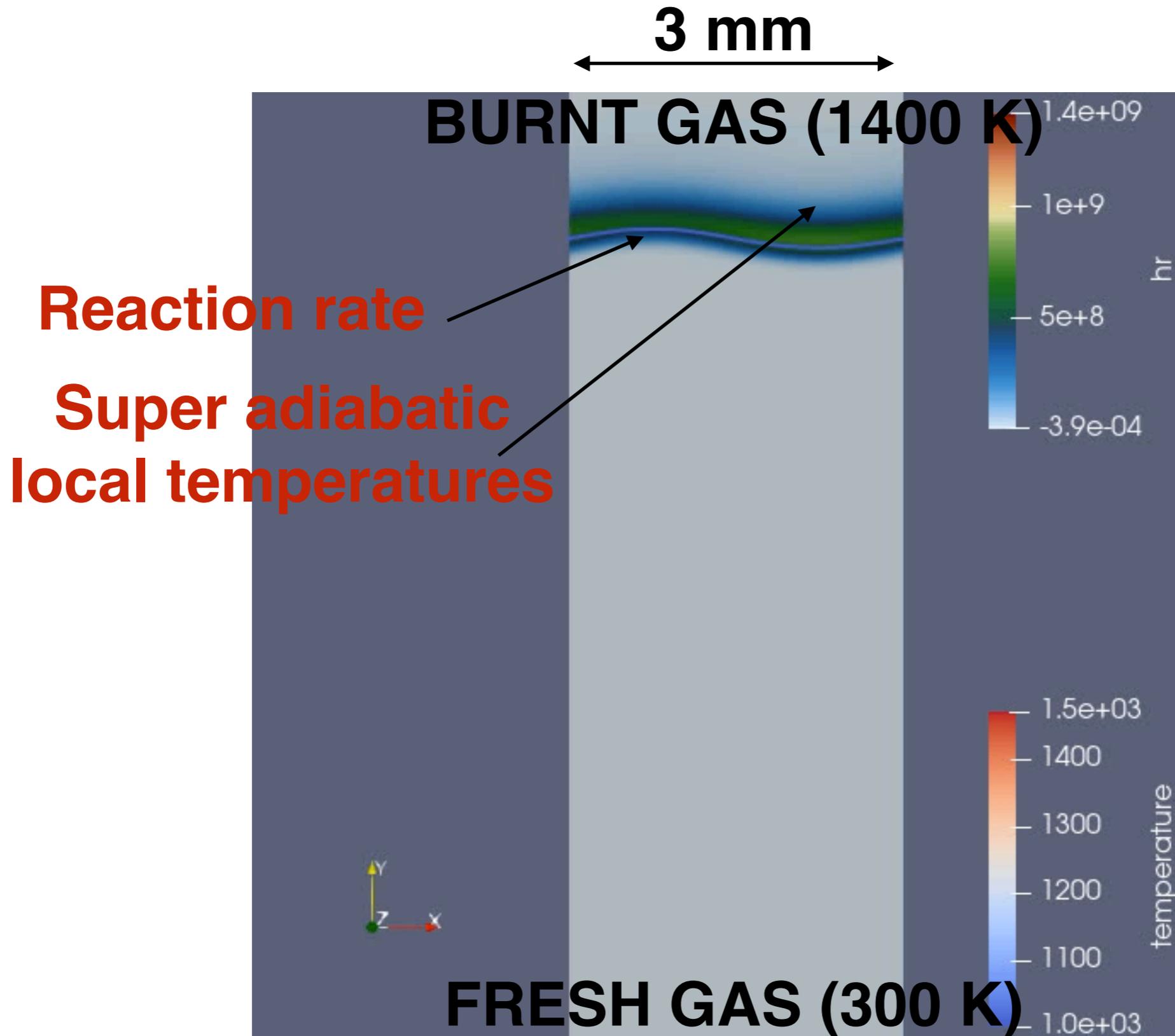
Main consequence of this fact:

‘thermodiffusive instabilities’





Flames which should be flat are not ... flat



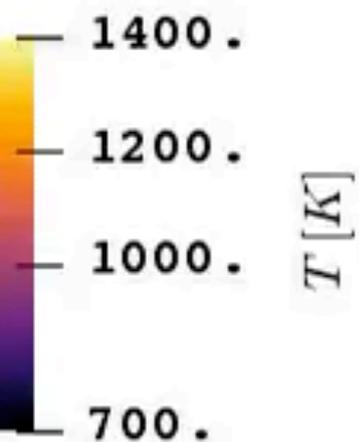
THERMODIFFUSIVE INSTABILITIES

THESE CELLS APPEAR IN
LAMINAR FLOWS.

WHAT HAPPENS IN A
TURBULENT FLOW ?

SEE DOUASBIN/JARAVEL
FOR THE LATEST NEWS THIS
AFTERNOON

***TD
stable***



T [K]



***TD
unstable***

th $Le=1$

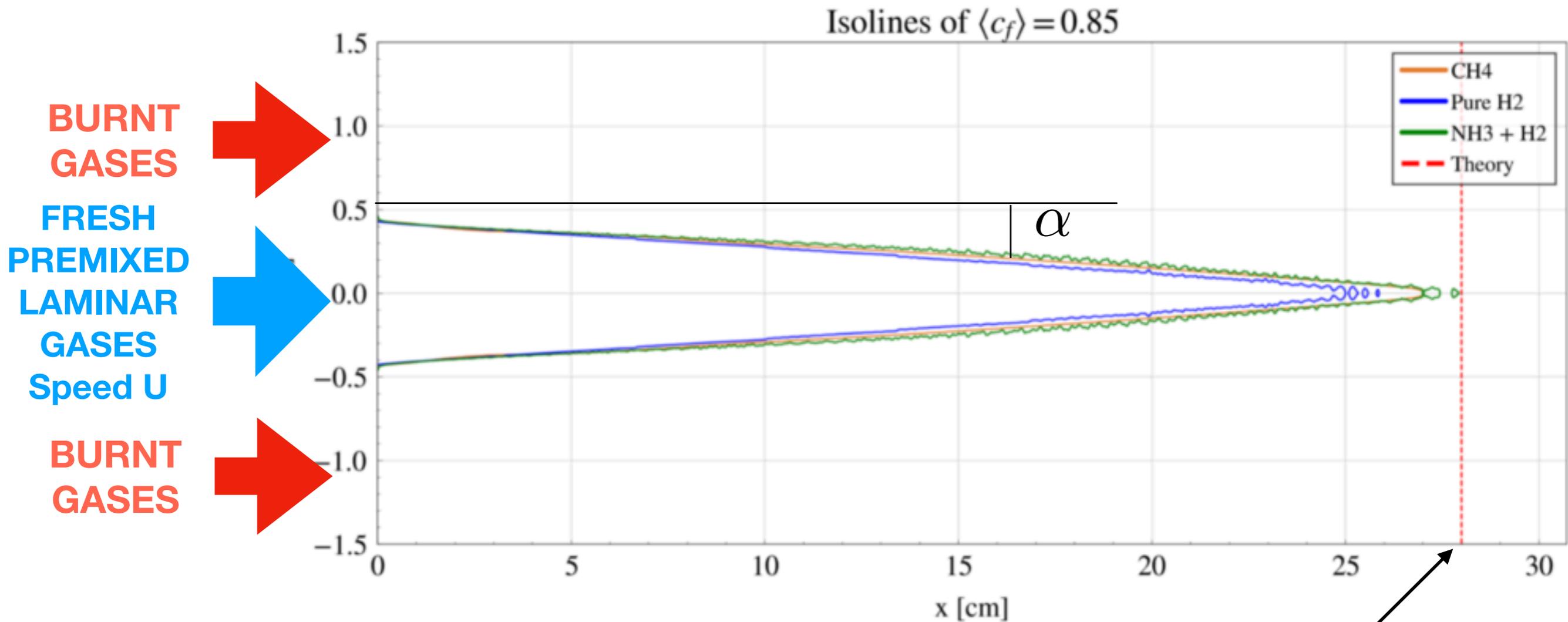
Real Le

In a turbulent case, TD can interact with flame/turbulence wrinkling and usual models wont work...

How do we know?: DNS (Attili, Berger, Pitsch or Howarth, Aspden)
Here we show results by CERFACS and NTNU (Gaucherand, Laera, Schulze-Netzer, Poinso, *Combustion and Flame*, 256, 2023, 112986)

- 3 premixed flames with exactly the **same laminar flame speed (0.38 m/s) and flame thickness**
 - Methane-air flame, $\phi = 1$
 - Hydrogen-air flame, $\phi = 0.45$
 - Ammonia/hydrogen flame, $\phi = 1$, $X_{NH_3} = 0.535$
- Chemistry:
 - CH4: 1-step mechanism with Pfitzner for reaction rate source term (5 species, 1 reaction)
 - NH3-H2: ARC mechanism from CRECK (14 species, 7 QSS, 174 reactions)
 - H2: detailed San Diego mechanism H2 (9 species, 42 reactions)

Laminar flames front: all the same, as expected

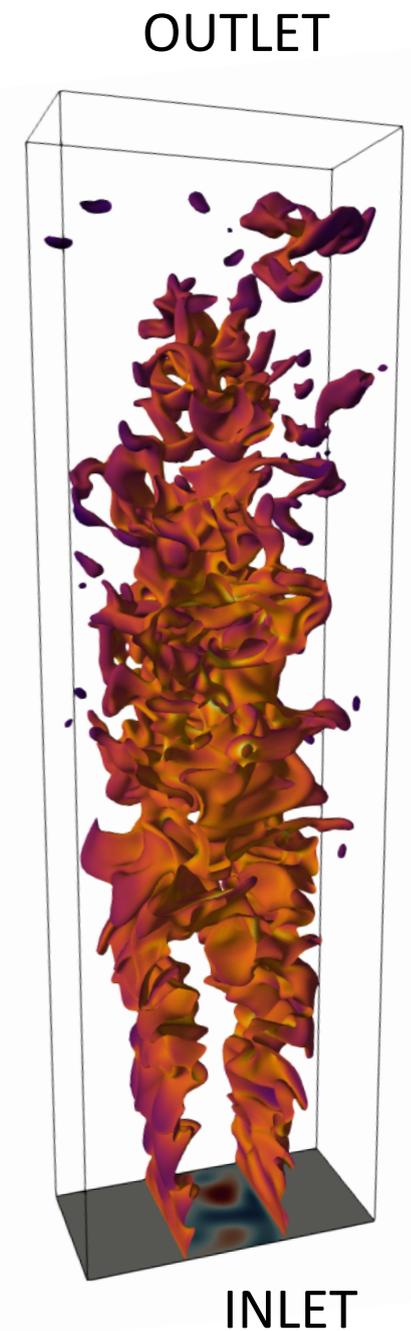


Theoretically, the flame length (given by $\sin(\alpha) = s_L/U$) is $L_f = 28 \text{ cm}$

IN A LAMINAR FLOW, EVERYTHING IS FINE

Numerical set up of DNS: turbulent cases

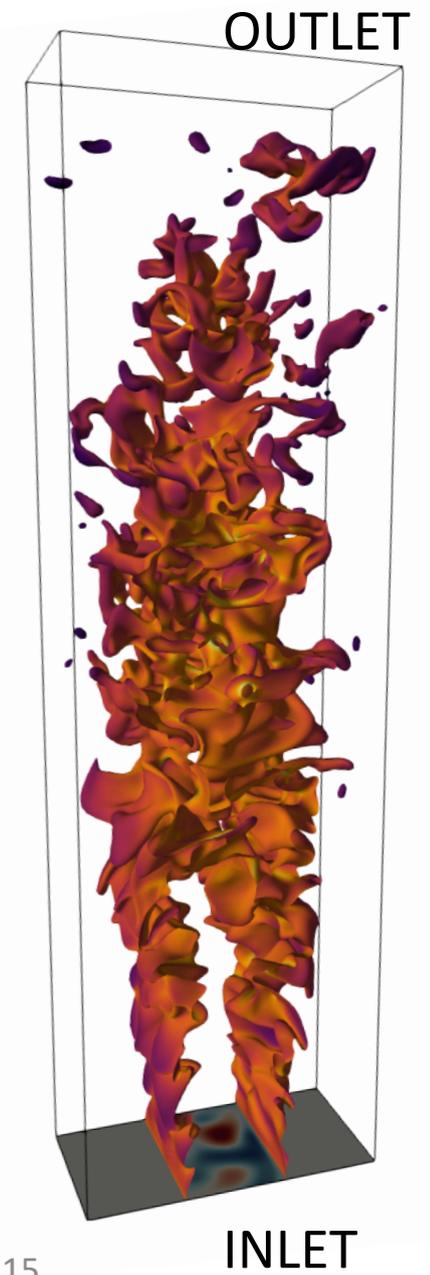
	CH4-air	H2-air	NH3/H2-air
Domain size Lx [cm]	10.24	5.12	8.96
Number of points in front flame [-]	7	11	
Number of points (Nx/Ny/Nz) [-]	1601/401/201	1281/641/321	2191/627/314
Simulation time [ms]	8	4.1	7.2



Numerical set up of DNS: turbulent cases

- For most turbulent combustion models, these 3 flames are similar because these models inputs are:
 - *Laminar flame speed s_L and thickness δ_L*
 - *Turbulence RMS speed u' and integral scale L*
- In other words, most existing models would be blind to the differences between these flames

Let us look at the DNS results:



CH4-air flame

Time: 0.000020



NH3/H2- air flame

Time: 0.000020

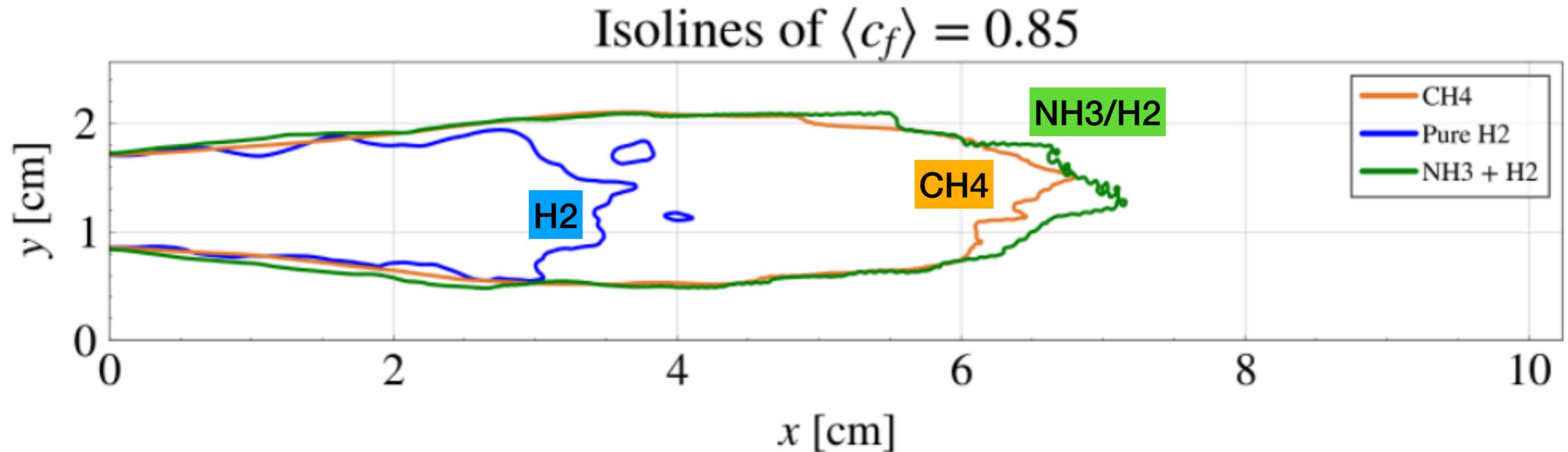


H2-air flame

Time: 0.000016

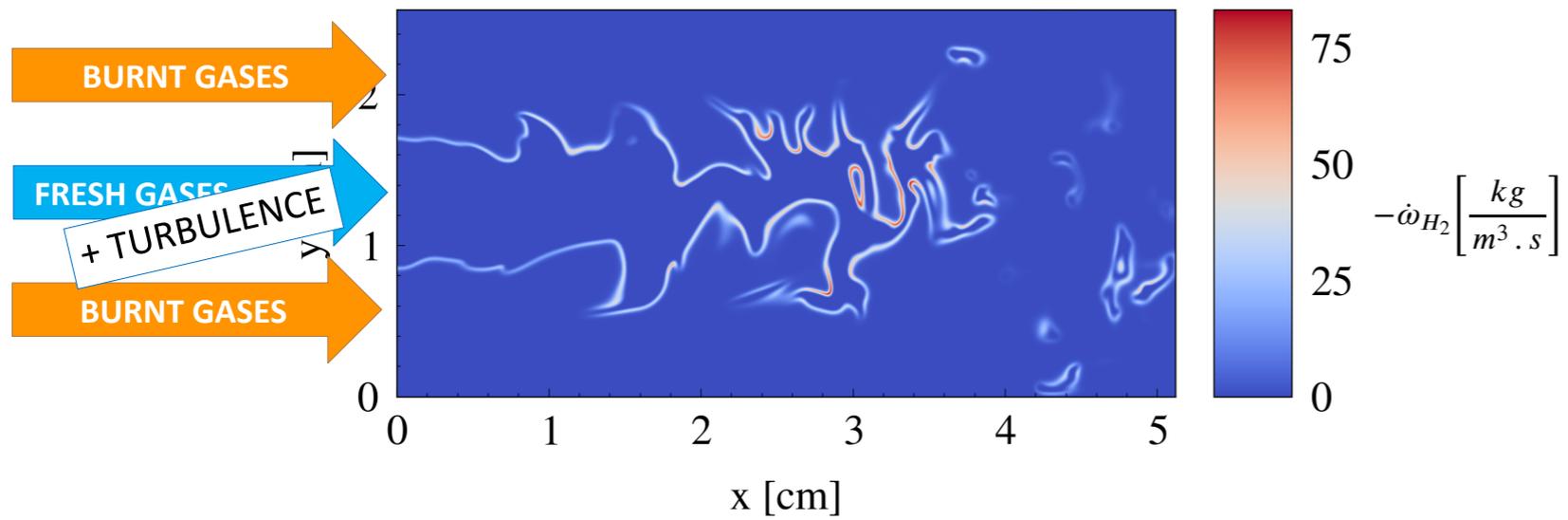
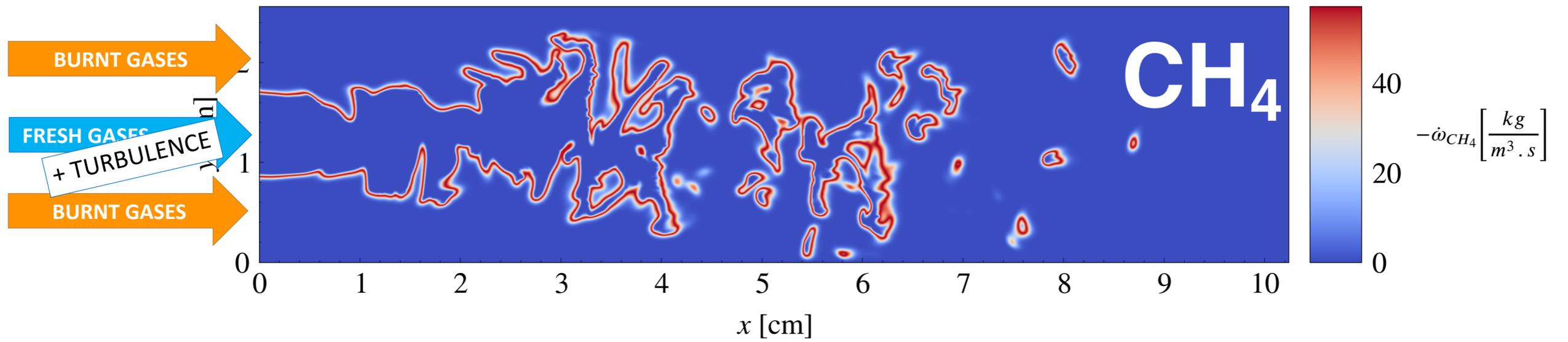


Turbulent flames: MEAN position

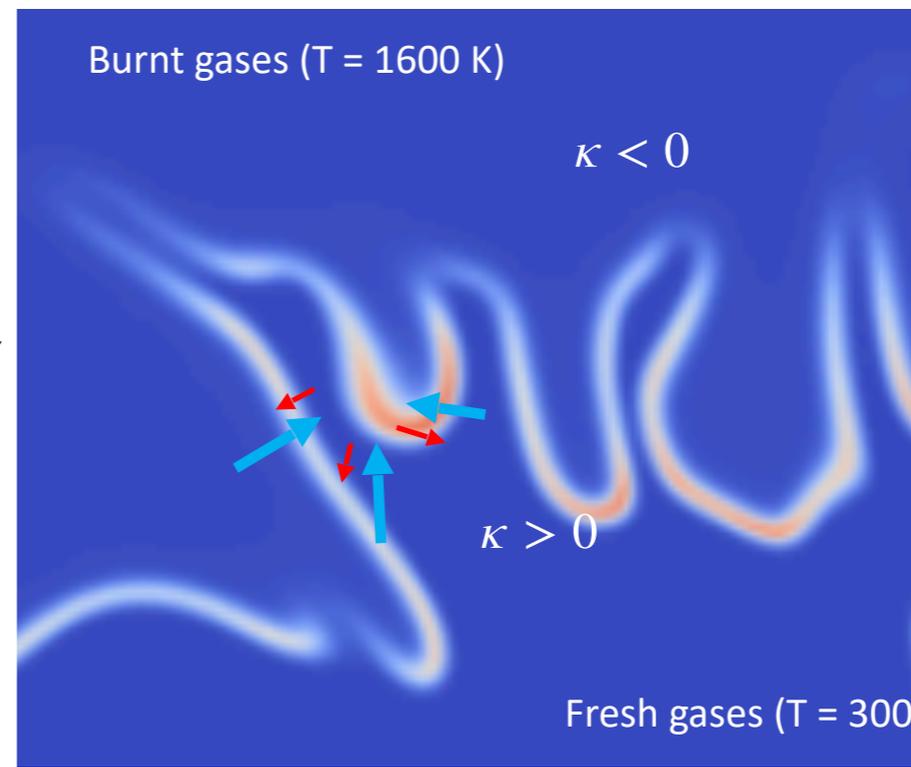
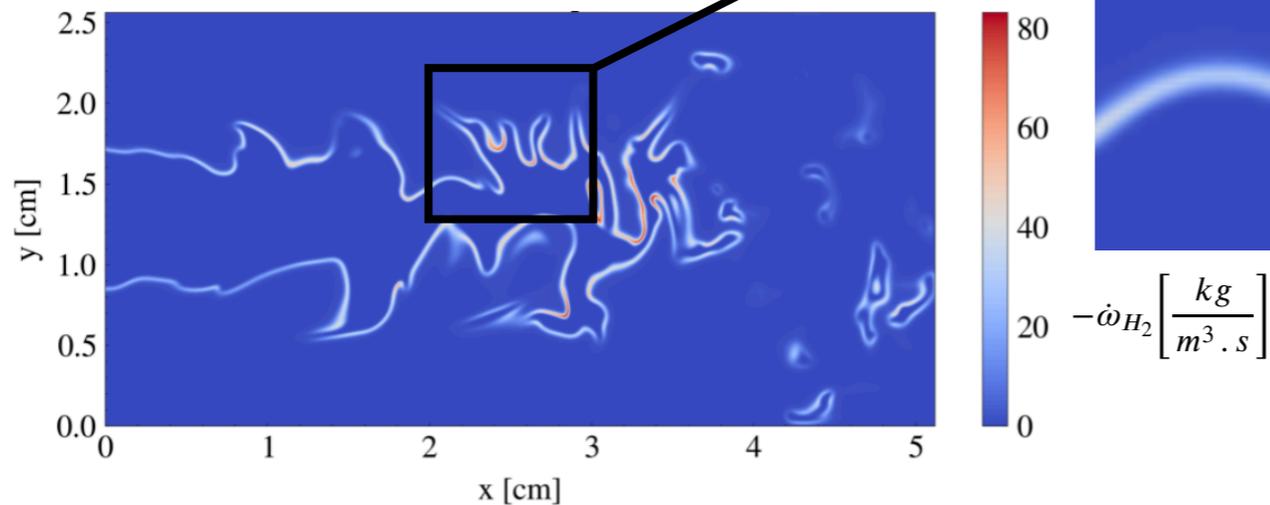


→ Pure H2 / AIR turbulent flame length almost 2 times less compared to CH4 and NH3-H2

→ Flame/turbulence interaction models using ‘only’ u'/s_L and L/δ_L as inputs will be wrong for H2

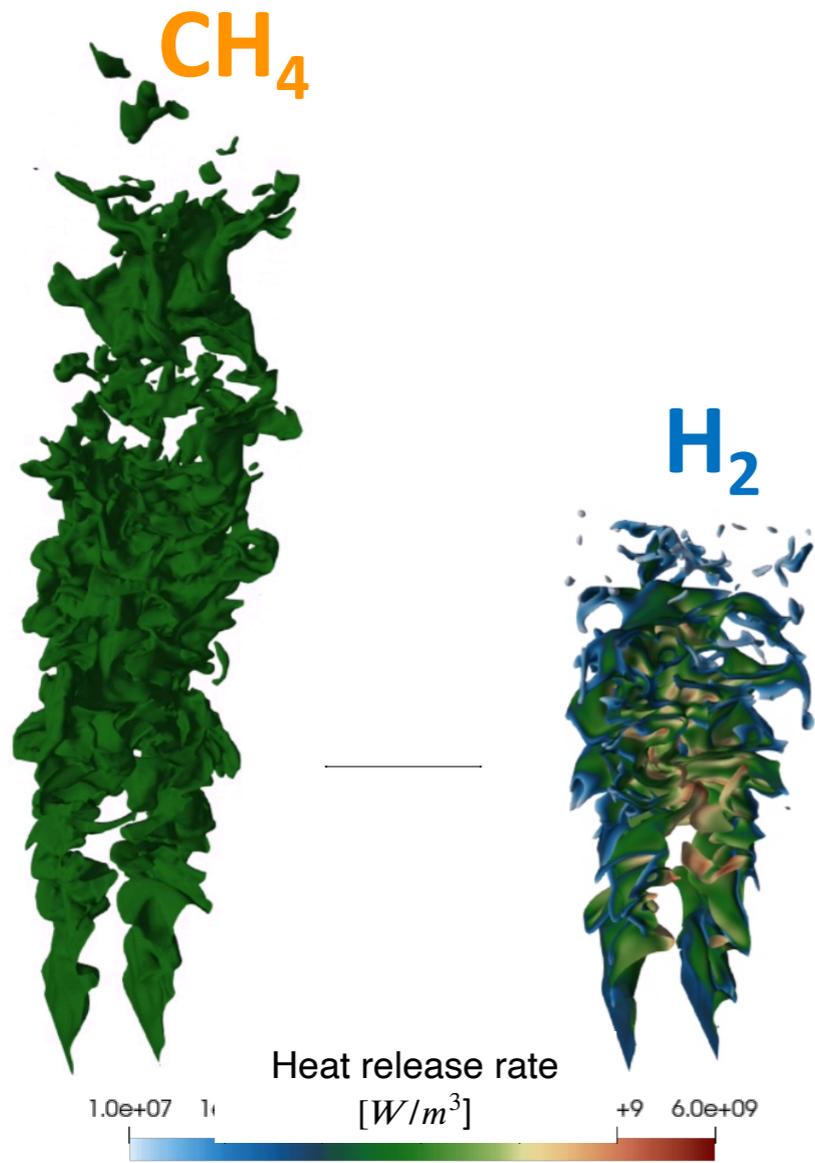


$$Le_{H_2} = \frac{\mathcal{D}_{th}}{\mathcal{D}_{H_2}} \approx 0.3 \quad (Le_{CH_4} \approx 1)$$



← H2 mass diffusivity
→ Thermal diffusivity

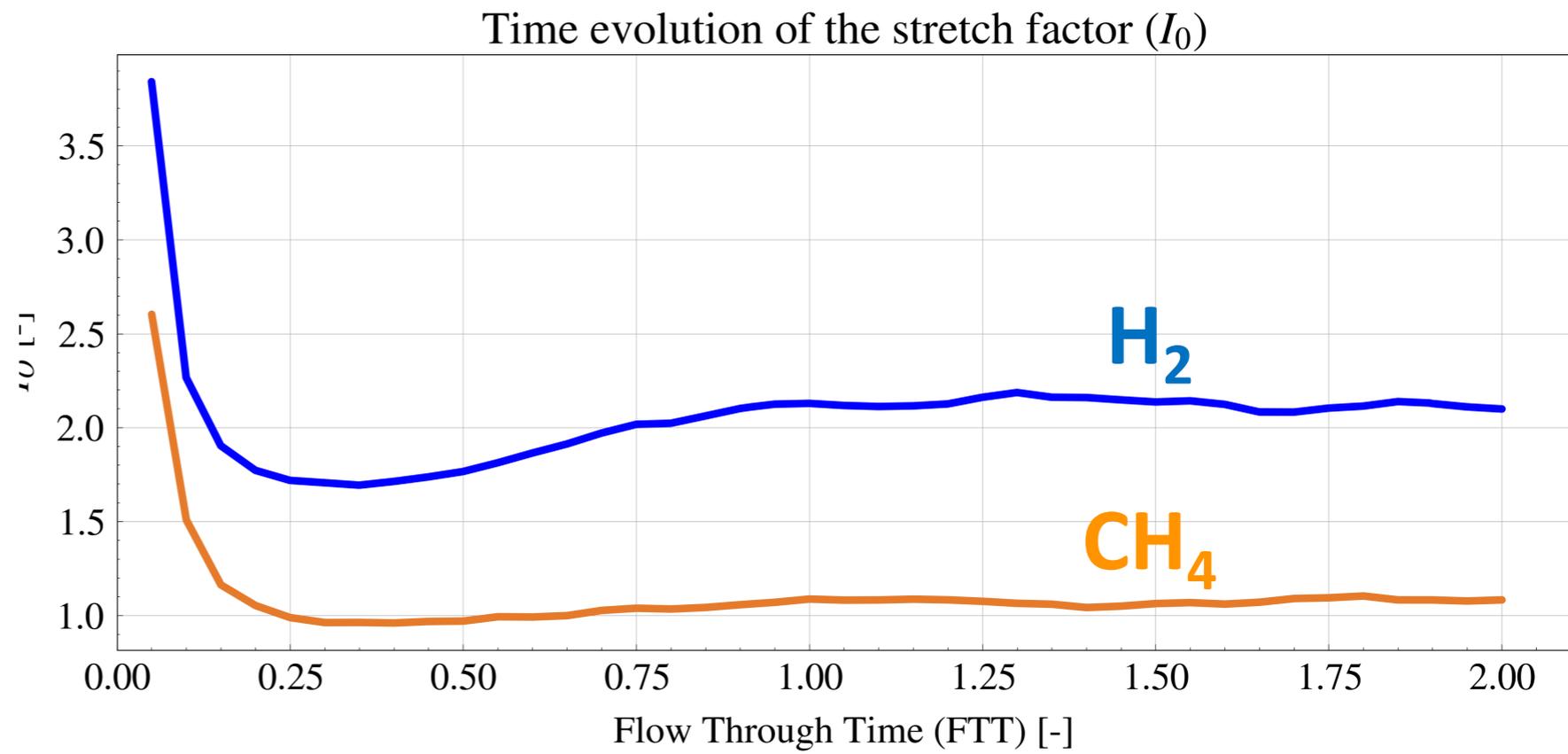
→ What your eyes show you - a strong effect of curvature- is actually misleading: the main effect is NOT that certain curved flame zones burn more than others. The main effect is that ALL flame zones burn much more on average than the laminar flame speed



Consumption speed averaged along the flame

$$I_0 = \frac{\bar{s}_c}{s_L^0}$$

Laminar flame speed

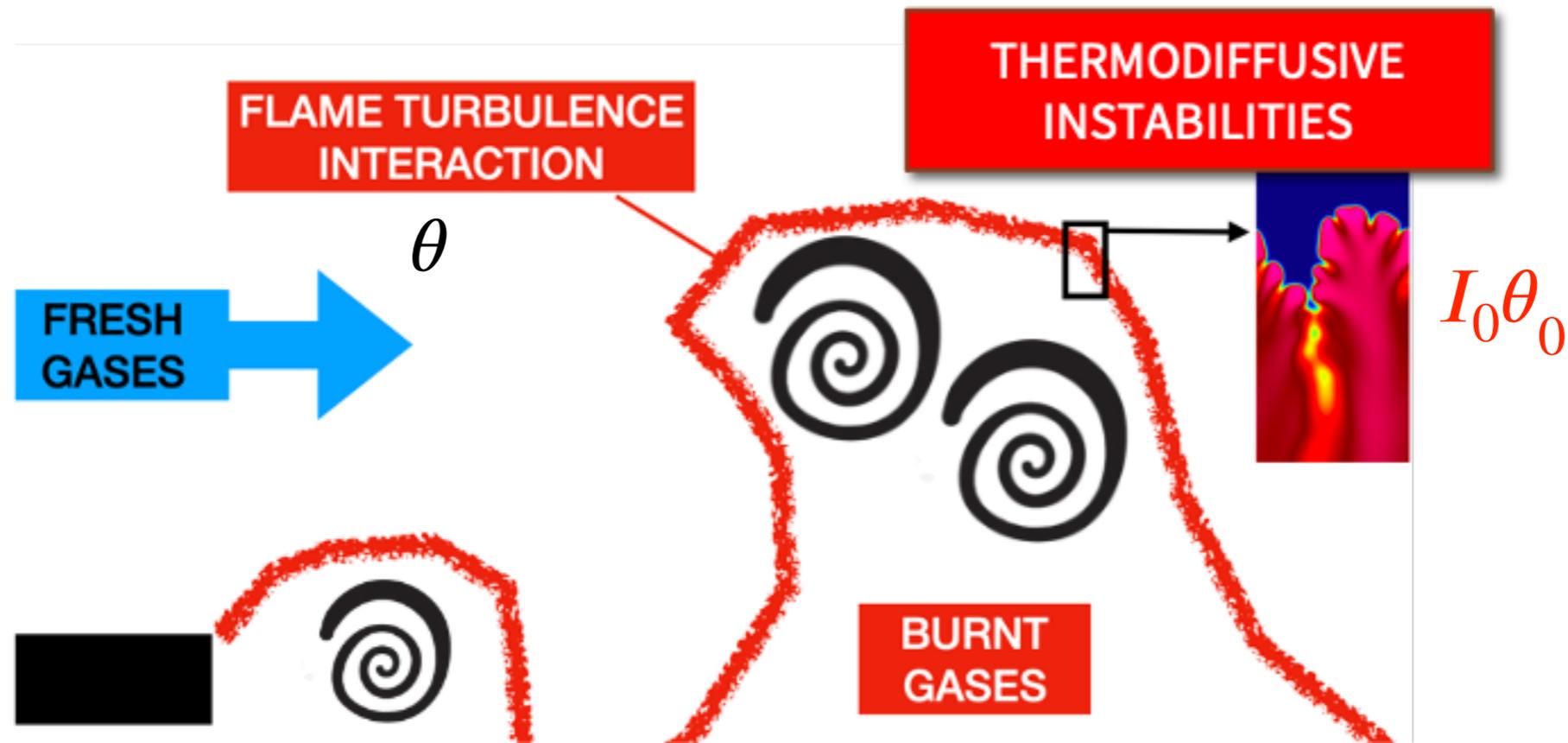


Introducing Thermodiffusive Effects into LES codes

- DNS tells us there is an issue with thermodiffusive effects for lean H₂ flames
- HOW DO WE FIX THIS IN LES (we can continue to do DNS but it does not give us a model we can use) ?
- One possible example: the TD TFLES models
 - Implemented in the usual Thickened Flame LES model: *A. Aniello, D. Laera, L. Berger, A. Attili and T. Poinsot. web.stanford.edu/group/ctr/ctrsp22/iv03_Aniello.pdf*
 - Using DNS data produced by Aachen (Dr. Lukas Berger, Dr. Xu Wen, Prof. Heinz Pitsch from Aachen University Dr. Antonio Attili, University of Edinburgh)
- Assumes scale separation:
 - Small: thermodiffusive effects are supposed to play a role only at very small scales, not seen by the LES.
 - Large: at the LES grid level, standard flame/turbulence models are used.



The TD TFLES model:



The turbulent flame speed s_T becomes:

$$s_T = s_L^0 \theta I_0 \theta_0$$

With:

s_L^0 : laminar flame speed

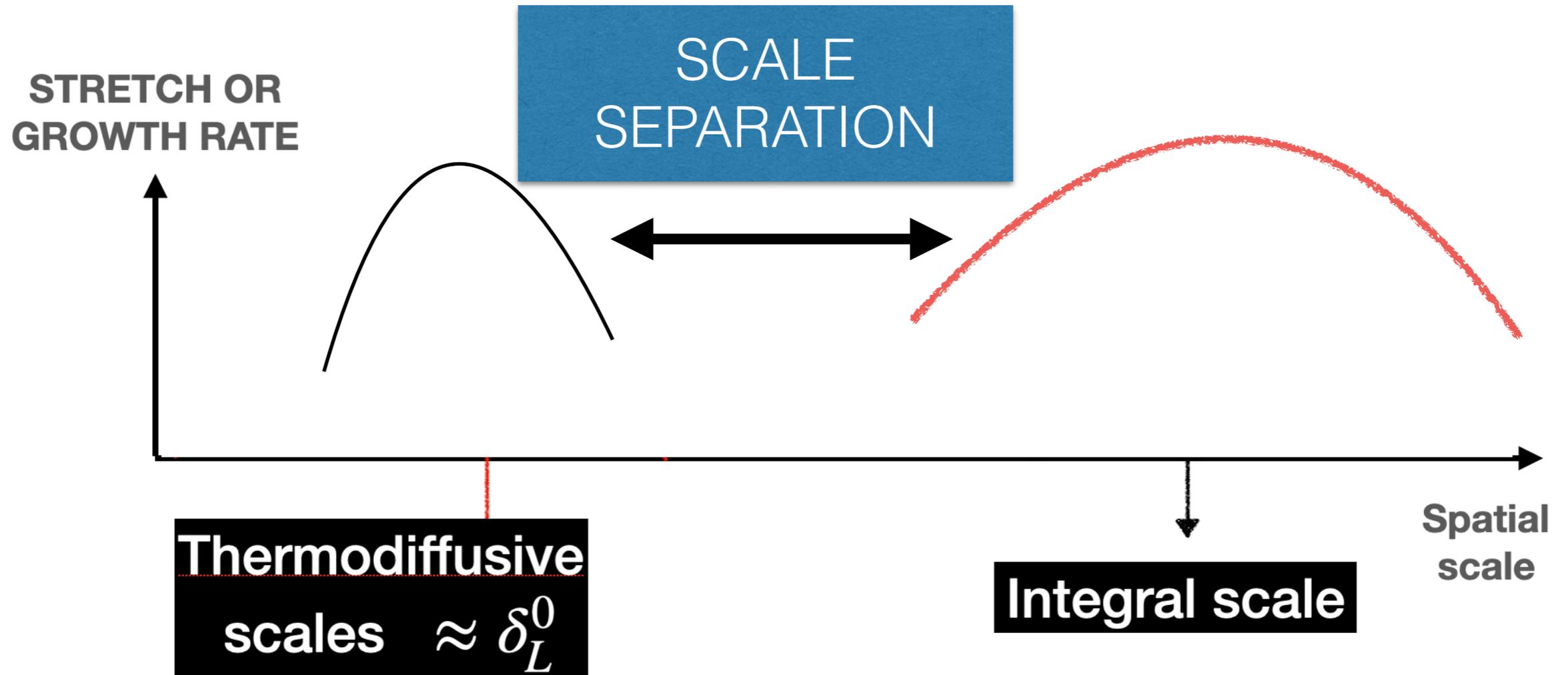
θ : wrinkling due to turbulence

I_0 : increase of consumption speed due to thermodiffusive instabilities.

θ_0 : wrinkling due to instabilities

STANDARD TFLES MODEL

TD TFLES EXTENSION



Thermodiffusive effects live only at small scales

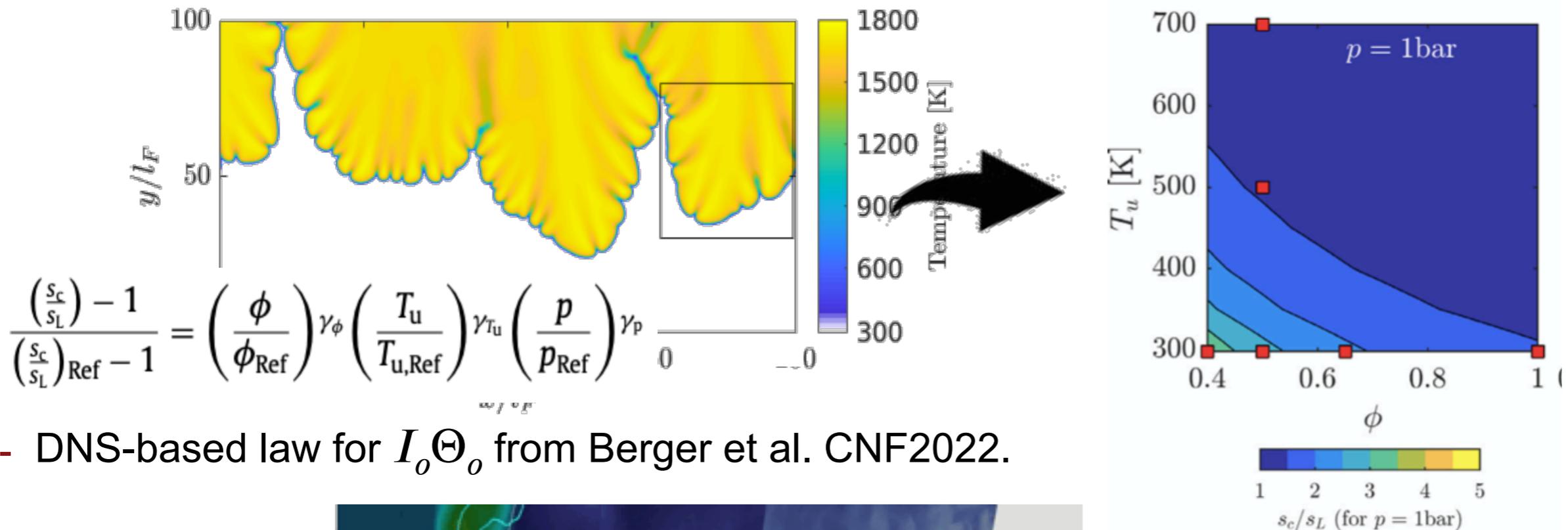
Modeled by increasing the local 'laminar' flame speed

Flame/turbulence interaction lives at large scales

Modeled by using the efficiency function of the TFLES model

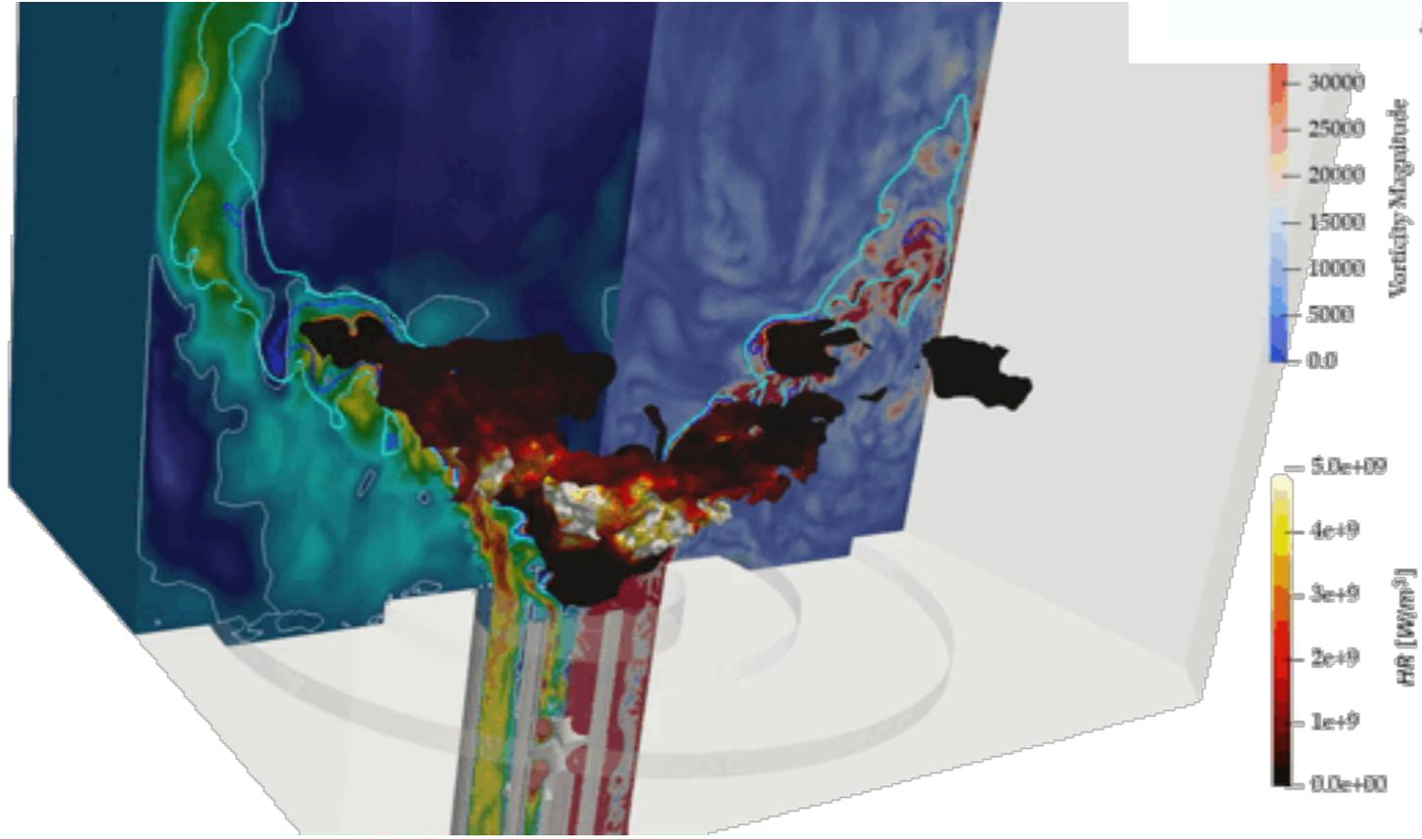
24 How do we obtain $I_0\Theta_0$? From DNS (Aachen group)

$I_0\theta_0$



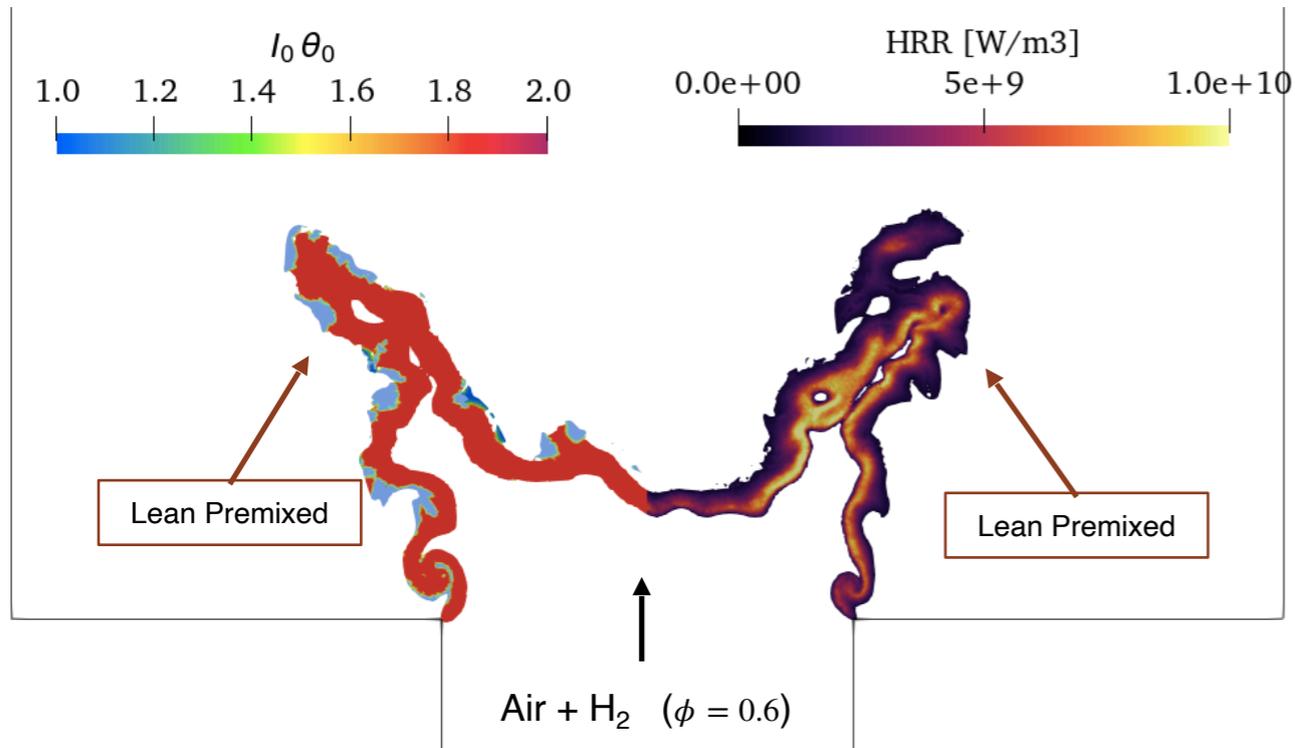
$$\frac{\left(\frac{s_c}{s_L}\right) - 1}{\left(\frac{s_c}{s_L}\right)_{\text{Ref}} - 1} = \left(\frac{\phi}{\phi_{\text{Ref}}}\right)^{\gamma_\phi} \left(\frac{T_u}{T_{u,\text{Ref}}}\right)^{\gamma_{T_u}} \left(\frac{p}{p_{\text{Ref}}}\right)^{\gamma_p}$$

- DNS-based law for $I_0\Theta_0$ from Berger et al. CNF2022.

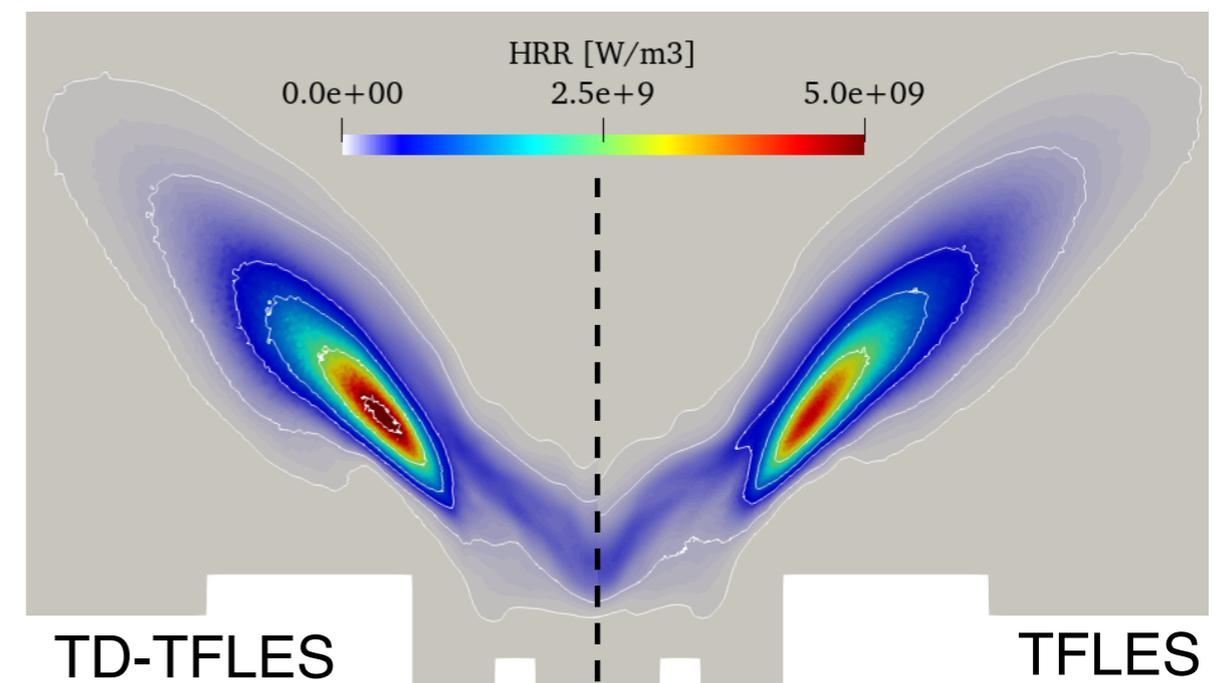
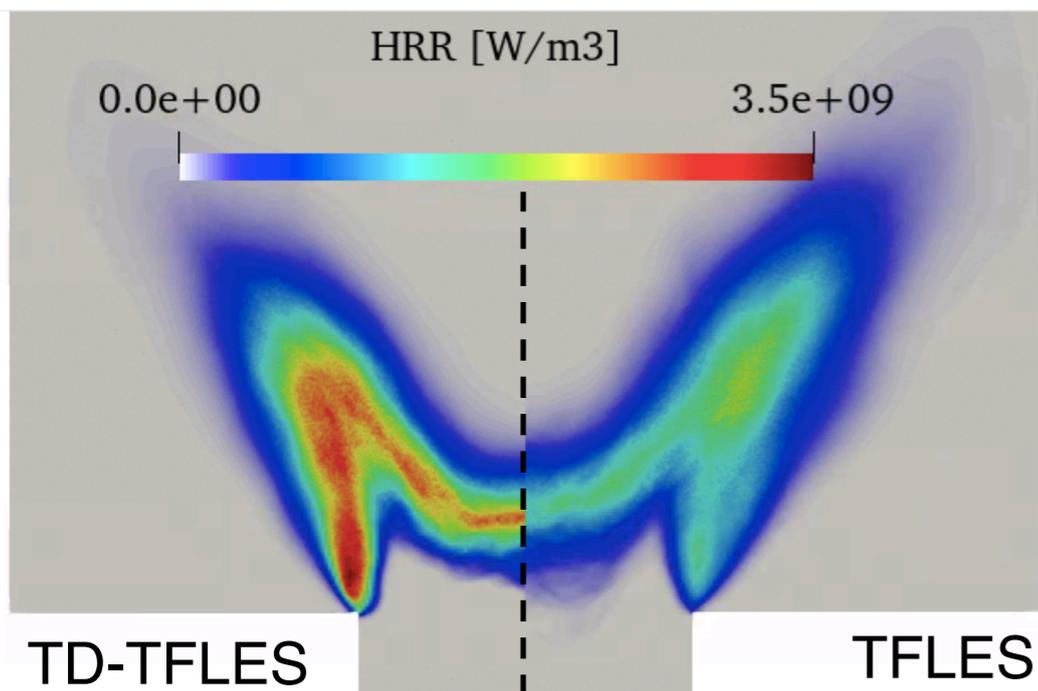
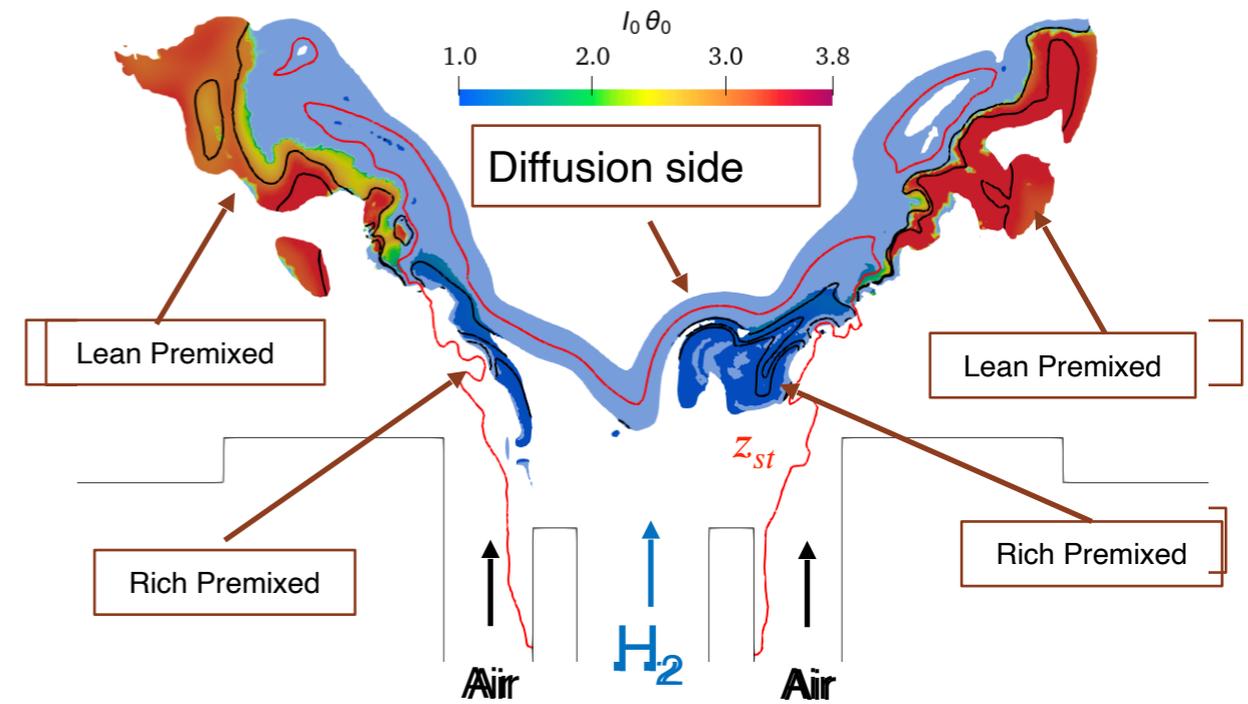


A posteriori test of TD LES for HYLON and TU Berlin flames

Fully premix lean flame @ TUB



Partially premix flame @ HYLON



Another CFD issue: walls and H2

Why worry about walls?

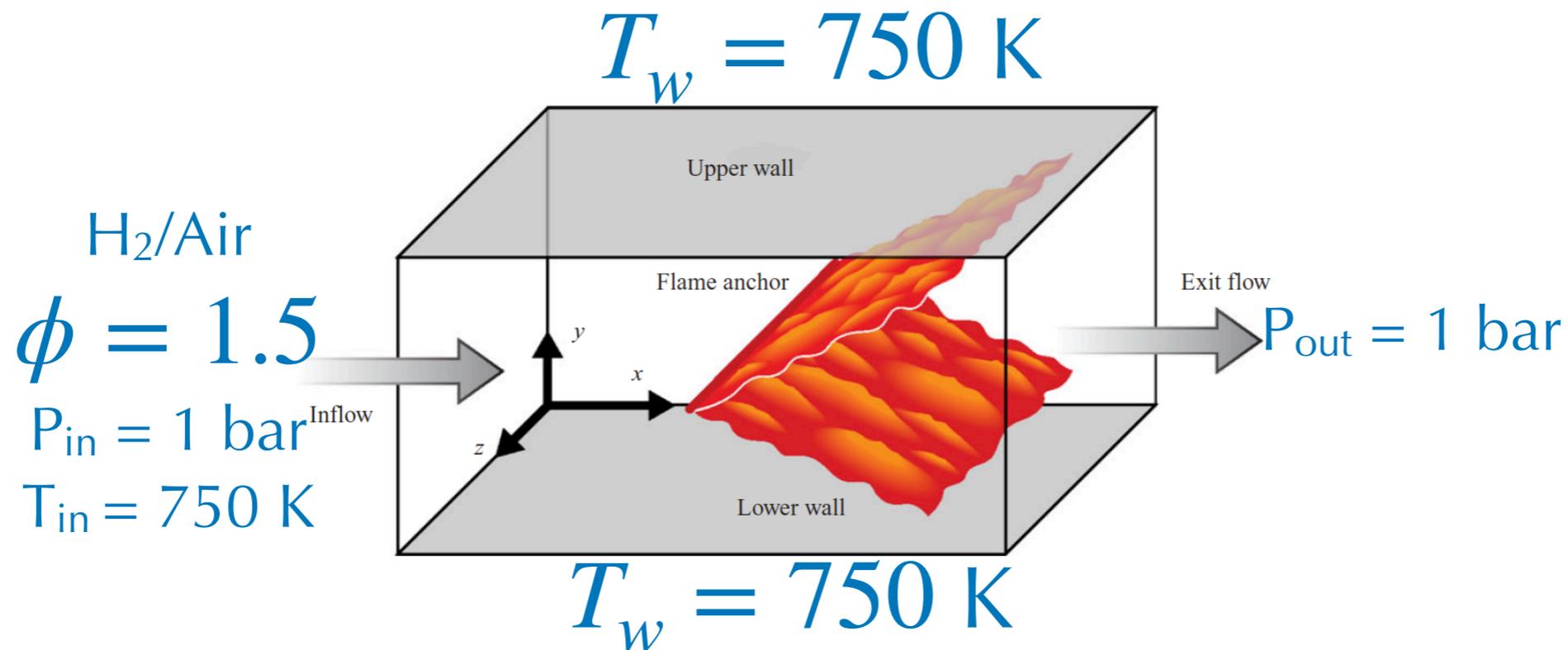
- ❖ The problem of the heat load generated by hydrogen flames is important because hydrogen flames come closer to walls
- ❖ For flames stabilized on injector lips: the temperature of the lips is an issue because (1) it controls the stabilization of the flame and (2) the lips should not burn and (3) even if they don't melt, hot walls might encourage flashback
- ❖ For flames impacting walls: the walls should not melt - > Safety issue
- ❖ In many processes (cement, iron, glass), we need to know the heat transfer to walls and the unpleasant possibility of catalytic reactions on the walls...

Reference: Loic De Nardi, Quentin Douasbin, Olivier Vermorel, Thierry Poinsot. Infinitely Fast Heterogeneous Catalysis Model for Premixed Hydrogen Flame-Wall Interaction. Comb. Flame. 261, 113328, 2024

SO: our CFD codes should predict flame-wall interaction correctly for H₂ flames. But they do not...



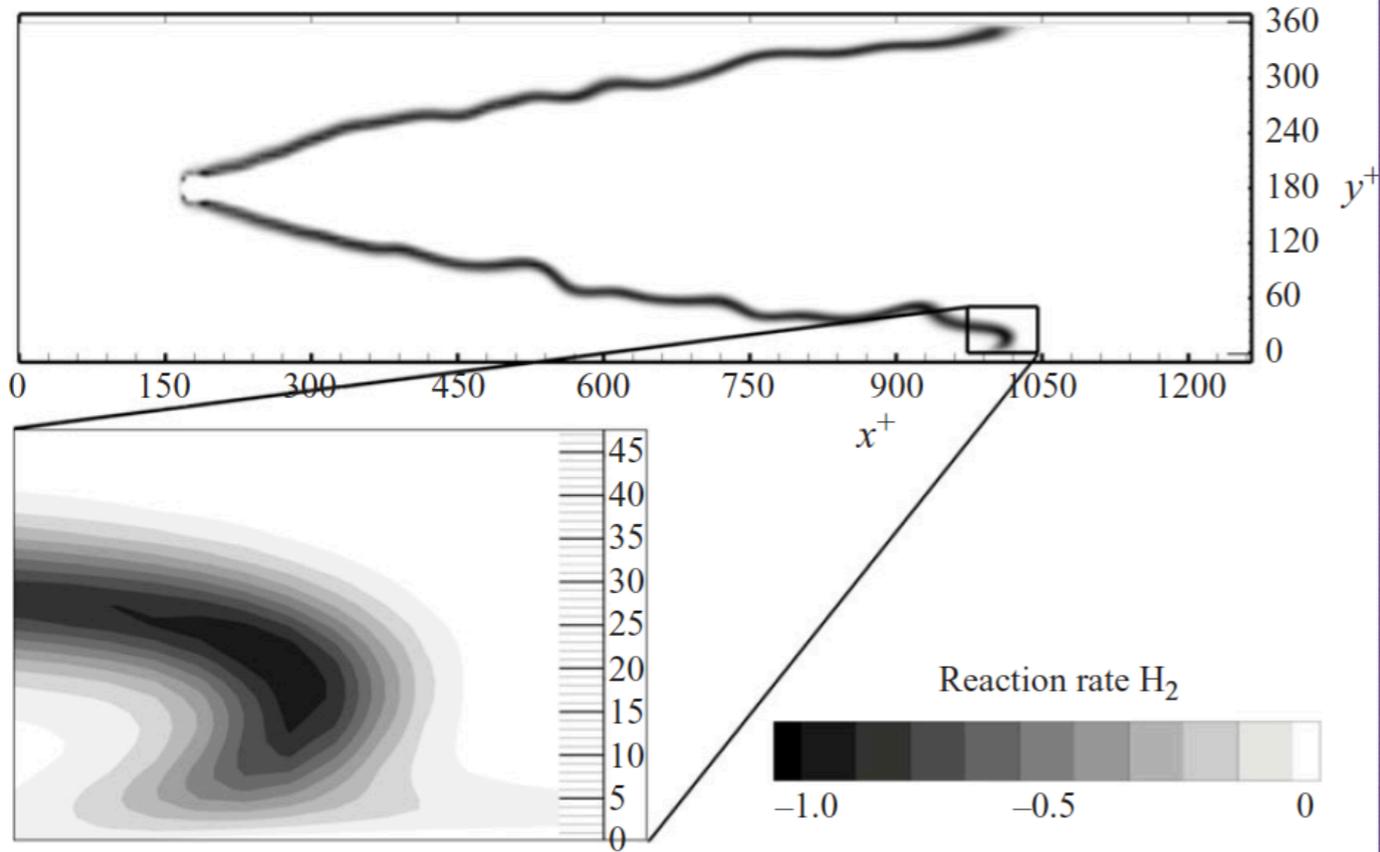
- ▶ A well-known DNS of H₂ FWI [6]



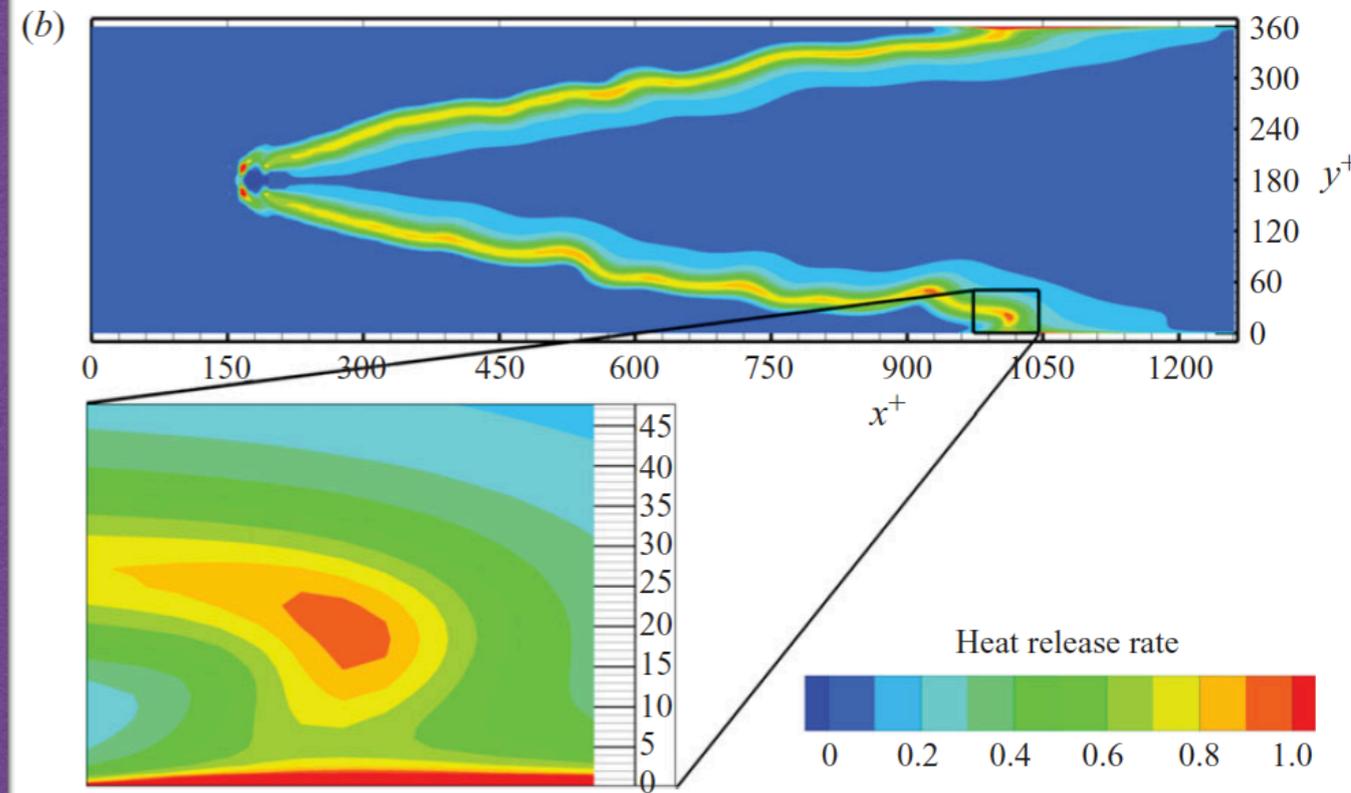
Laminar FWI - standard wall treatment



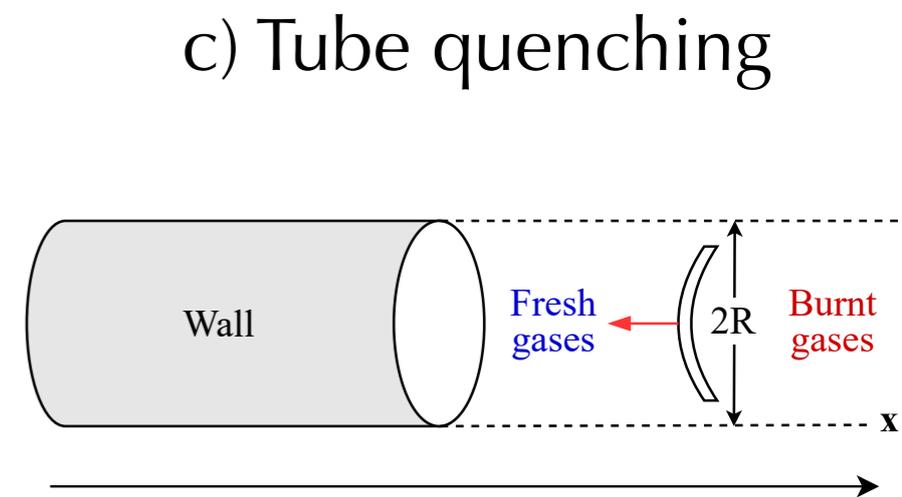
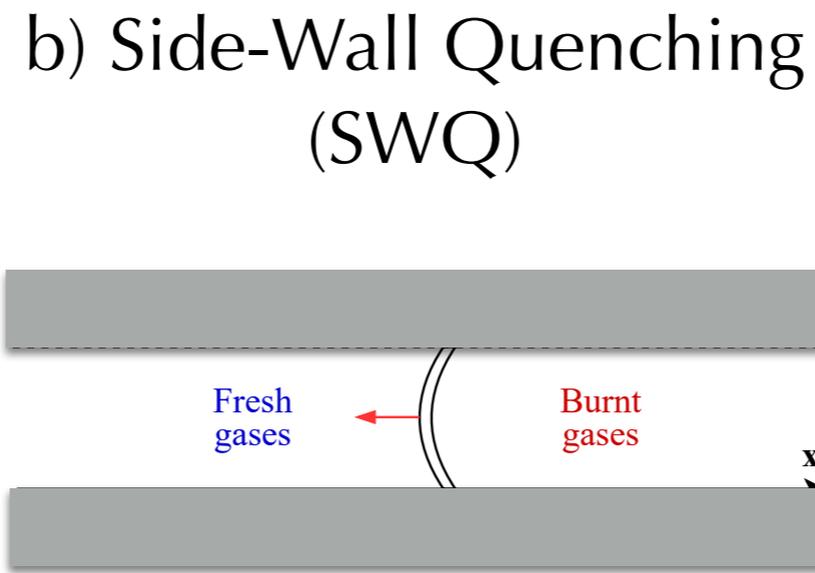
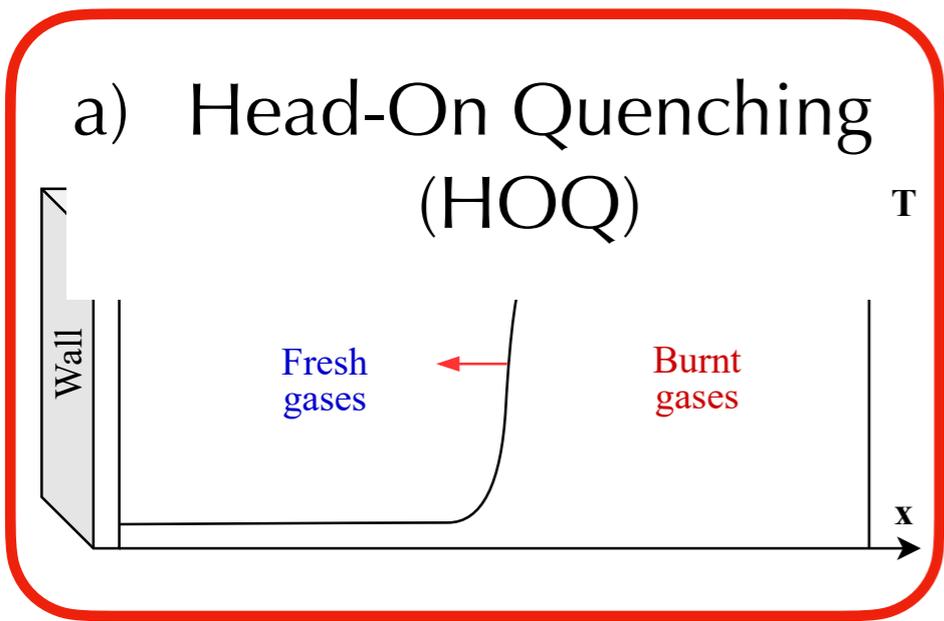
$\dot{\omega}_{H_2}$ is correct at the wall



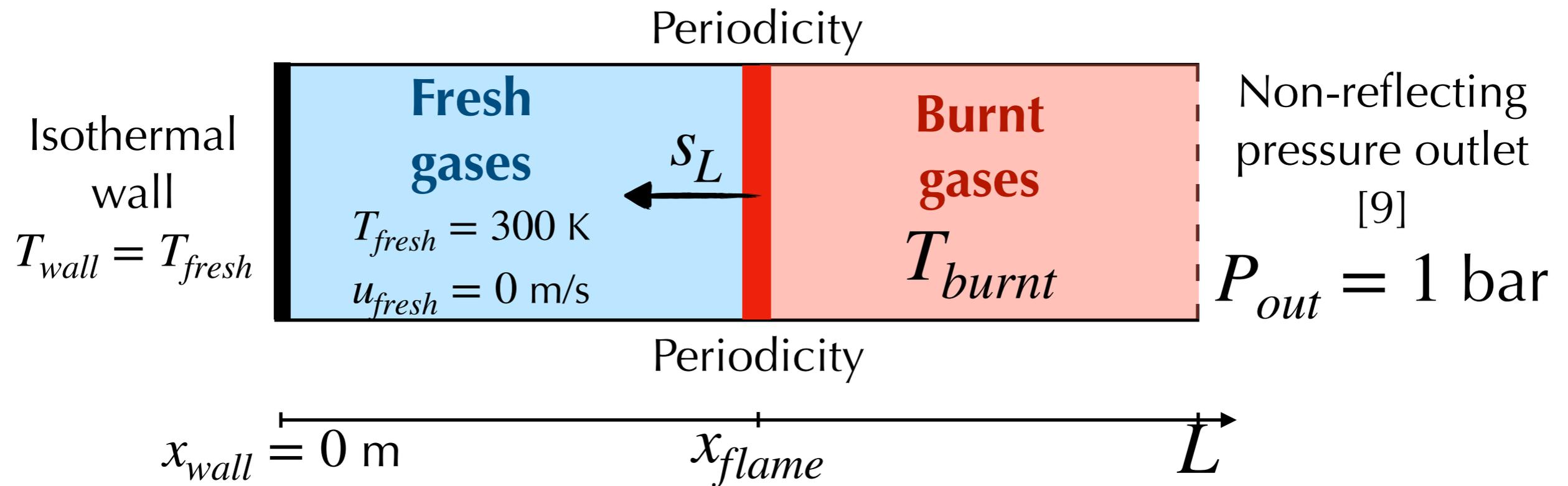
But $\dot{\omega}_T$ is not



How can we investigate this question simply?: we chose HOQ (Head On quenching) runs [7]



► 1D HOQ simulation with AVBP [8]

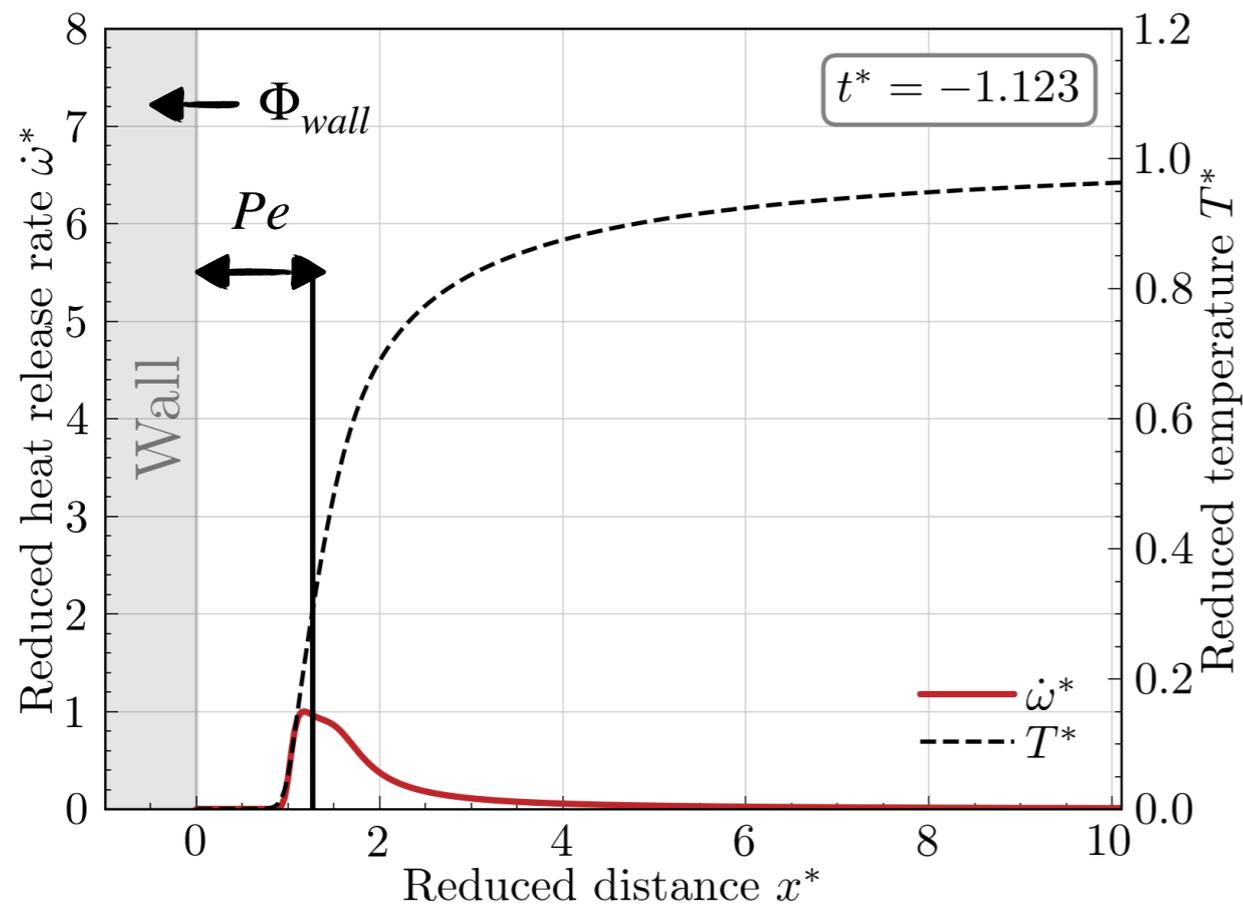


[8] Schönfeld and Rudgyard (AIAA, 1999)

[9] Poinso and Lele (JCP, 1993)



► Definitions



► Reduced variables

$$x^* = x/\delta_L$$

$$t^* = (t - t_Q)s_L/\delta_L$$

$$T^* = (T - T_f)/(T_b - T_f)$$

$$\dot{\omega}^* = \dot{\omega}_T/\dot{\omega}_T^0$$

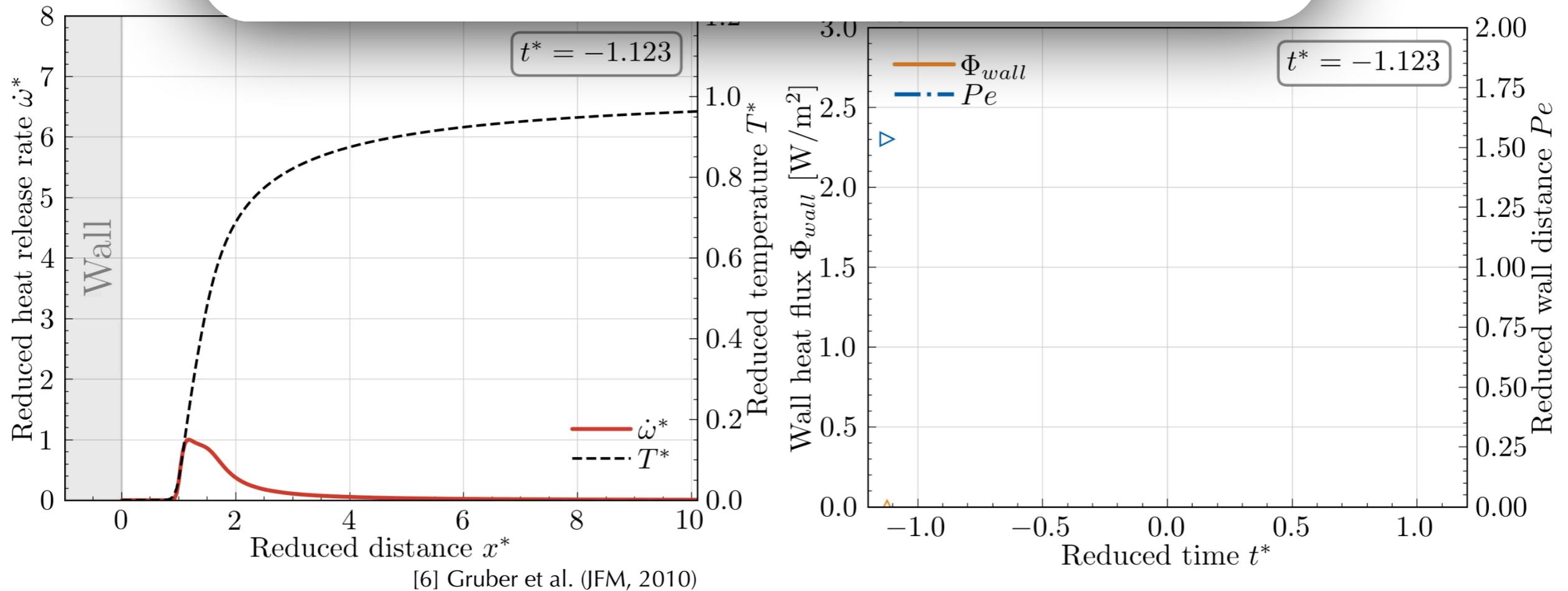
► Quantities of interest

$$Pe = x_{flame}/\delta_L$$

$$\Phi_{wall}$$

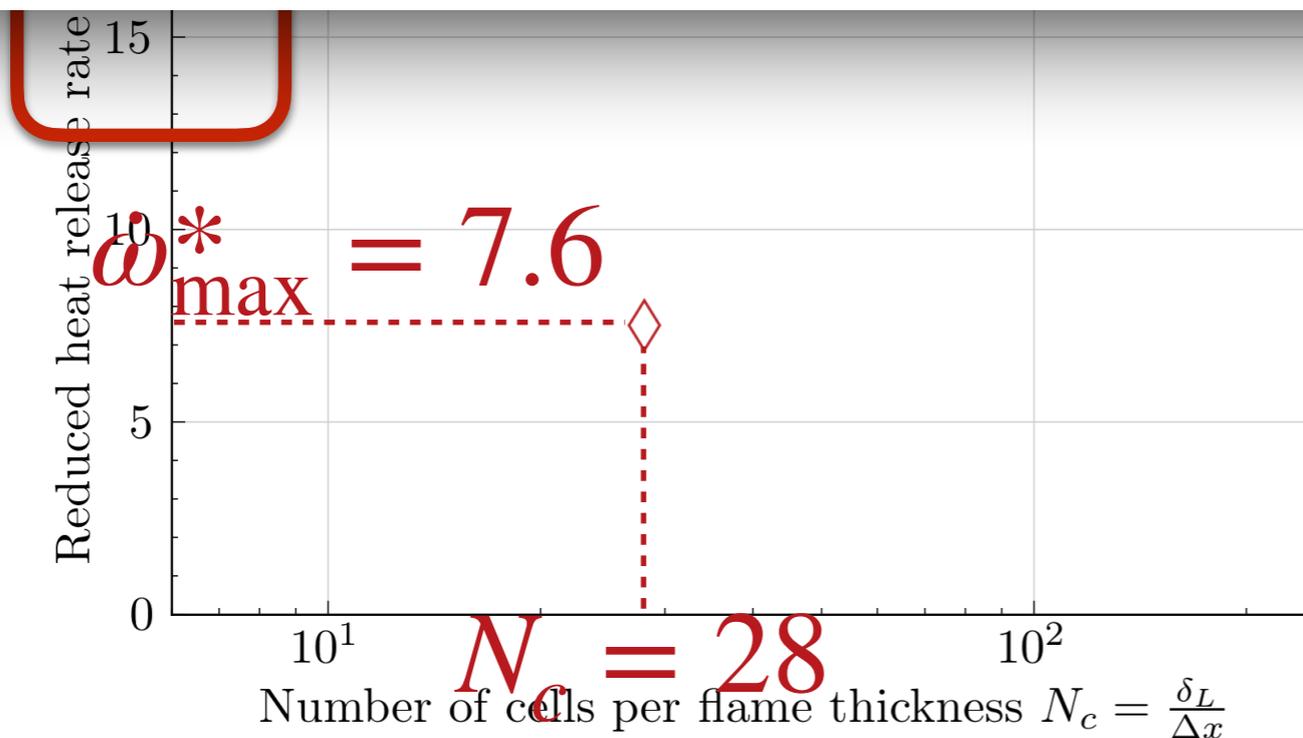


- Same spiked value of heat release $\dot{\omega}^*$ at the wall at $t^* = 0$ as in [6]
- Also noticed in other studies for H_2 [10-14] and CH_4 [15]



- State-of-the-art schemes [16-19] not fitted for FWI
- Tried most chemical schemes and found the same result
- Who is responsible: the H atom

ns!



- 1D HOQ performed for $N_c \in [7; 224]$
- $\dot{\omega}^*$ peak at quenching does not converge when the grid is refined

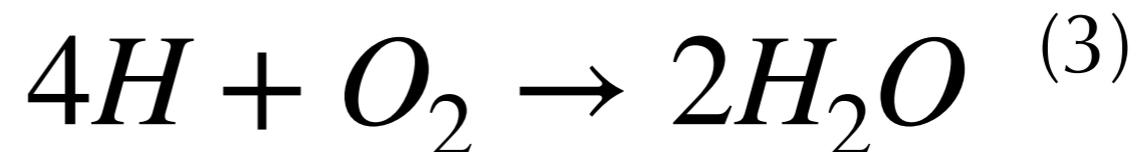
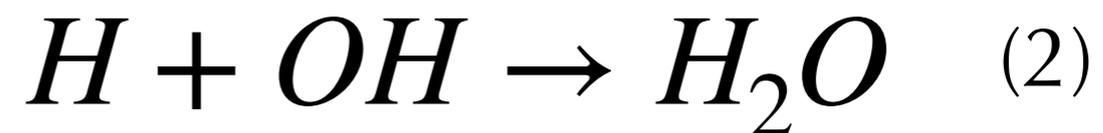
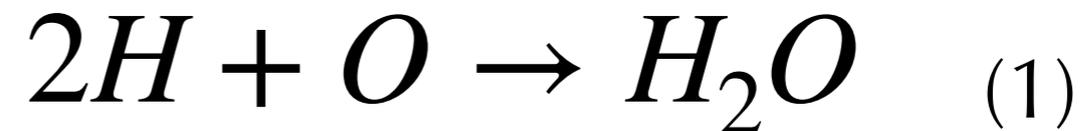
Who is responsible ?

ID	Reaction	k [cm ³ /mol/s/K ^{β}]	β [-]	E_a [cal/mol]
R1	$\text{H} + \text{O}_2 \rightleftharpoons \text{OH} + \text{O}$	3.52e+16	-0.70	17070
R2	$\text{H}_2 + \text{O} \rightleftharpoons \text{OH} + \text{H}$	5.06e+04	2.67	6291
R3	$\text{H}_2 + \text{OH} \rightleftharpoons \text{H}_2\text{O} + \text{H}$	1.17e+09	1.30	3635
R4	$\text{H}_2\text{O} + \text{O} \rightleftharpoons 2 \text{OH}$	7.60e+00	3.84	12780
R5	$2 \text{H} + \text{M} \rightleftharpoons \text{H}_2 + \text{M}^*$	1.30e+18	-1.00	0
R6	$\text{H} + \text{OH} + \text{M} \rightleftharpoons \text{H}_2\text{O} + \text{M}^*$	4.00e+22	-2.00	0
R7	$2 \text{O} + \text{M} \rightleftharpoons \text{O}_2 + \text{M}^*$	6.17e+22	-0.50	0
R8	$\text{H} + \text{OH} + \text{M} \rightleftharpoons \text{OH} + \text{M}^*$	4.71e+18	-1.00	0
R9	$\text{O} + \text{OH} + \text{M} \rightleftharpoons \text{HO}_2 + \text{M}^*$	8.00e+15	0.00	0
R10	$\text{H} + \text{O}_2 + \text{M} \rightleftharpoons \text{HO}_2 + \text{M}^\ddagger$	$k_\infty = 4.65\text{e}+12$ $k_0 = 5.75\text{e}+19$	0.44 -1.40	0.0 0.0
R11	$\text{HO}_2 + \text{H} \rightleftharpoons 2 \text{OH}$	7.08e+13	0.00	295
R12	$\text{HO}_2 + \text{H} \rightleftharpoons \text{H}_2 + \text{O}_2$	1.66e+13	0.00	823
R13	$\text{HO}_2 + \text{H} \rightleftharpoons \text{H}_2\text{O} + \text{O}$	3.10e+13	0.00	1721
R14	$\text{HO}_2 + \text{O} \rightleftharpoons \text{OH} + \text{O}_2$	2.00e+13	0.00	0
The H radical at the wall is the bad guy				
		$k_0 = 2.30\text{e}+18$	-0.90	-1702
R17	$2 \text{HO}_2 \rightleftharpoons \text{H}_2\text{O}_2 + \text{O}_2$	3.02e+13	0.00	1386
R18	$\text{H}_2\text{O}_2 + \text{H} \rightleftharpoons \text{HO}_2 + \text{H}_2$	2.30e+13	0.00	7950
R19	$\text{H}_2\text{O}_2 + \text{H} \rightleftharpoons \text{H}_2\text{O} + \text{OH}$	1.00e+13	0.00	3585
R20	$\text{H}_2\text{O}_2 + \text{OH} \rightleftharpoons \text{H}_2\text{O} + \text{HO}_2$	7.08e+12	0.00	1434
R21	$\text{H}_2\text{O}_2 + \text{O} \rightleftharpoons \text{HO}_2 + \text{OH}$	9.63e+06	2.00	3991

IFHC model for H₂ FWI

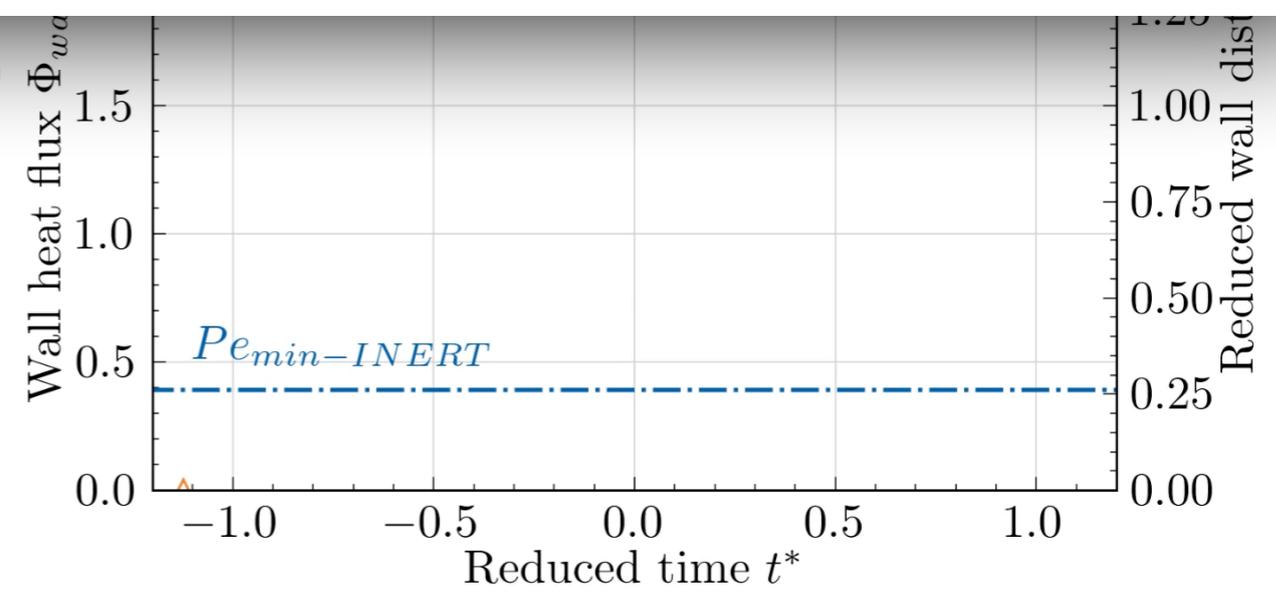
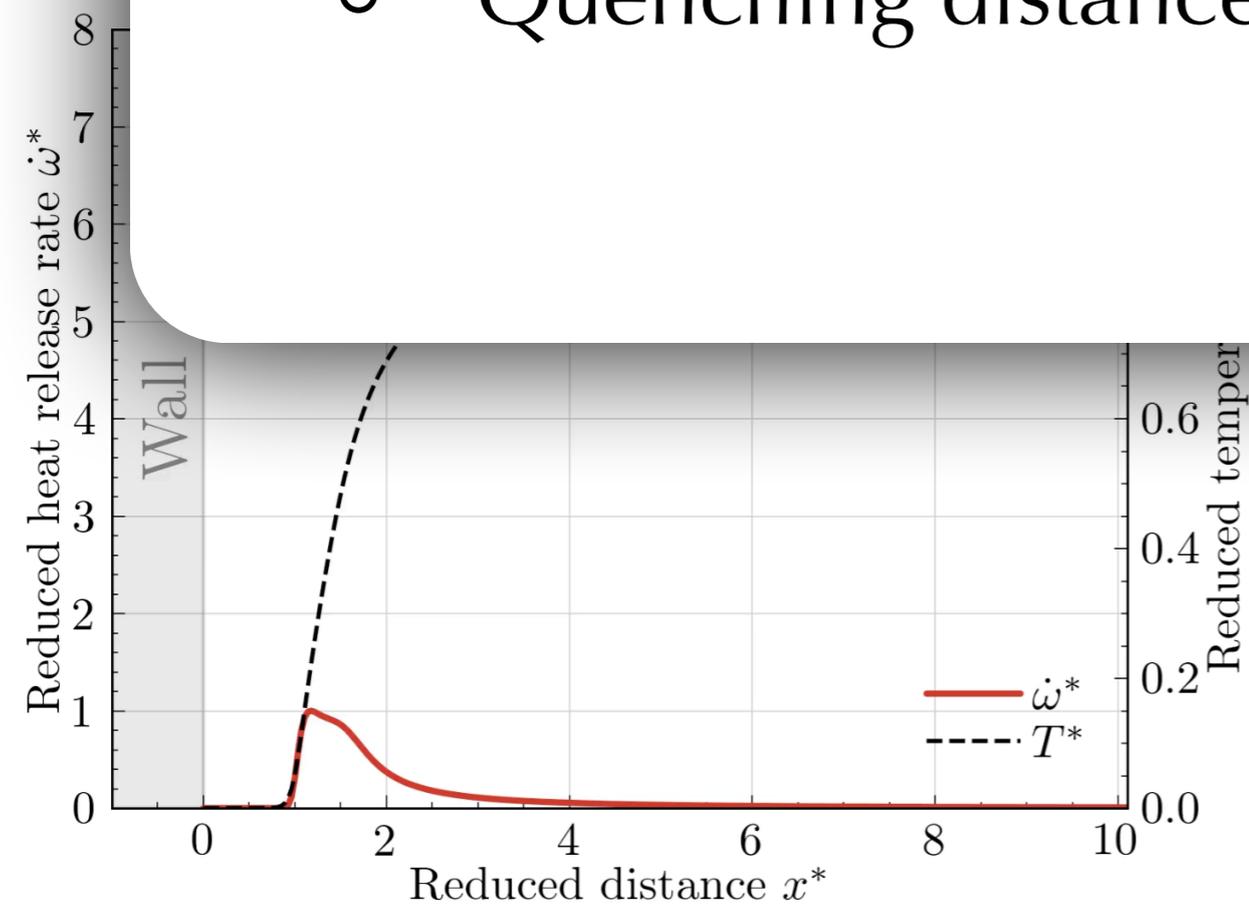
A simplified approach for FWI

- ▶ **There should be zero H radical at the wall surface: they disappear through catalytic reactions with the solid surface**
- ▶ **Infinitely Fast Heterogeneous Catalysis (IFHC) [3]**
 - Global surface chemistry: total, irreversible, occurring in 1 timestep:

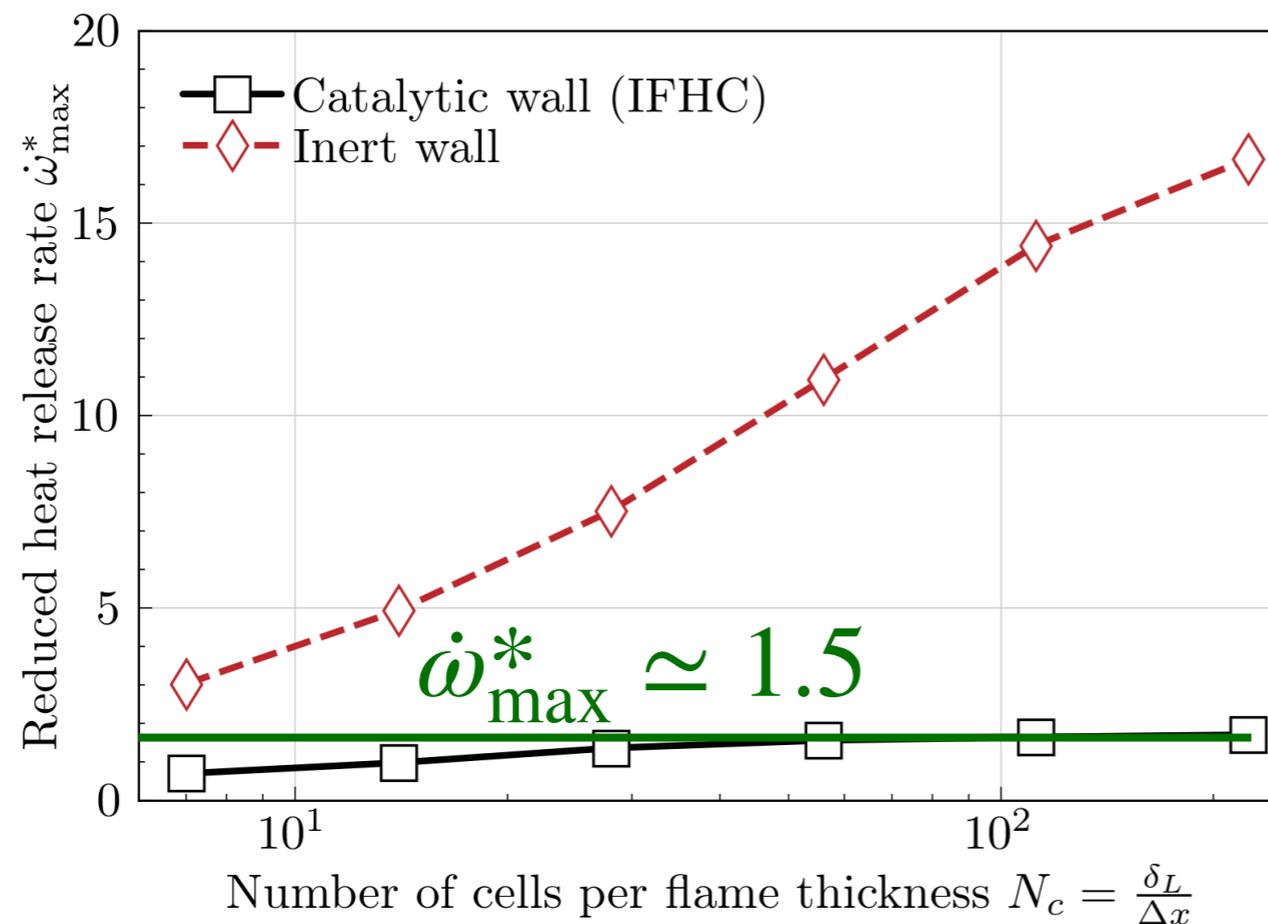




- No crazy heat release rate at the wall with IFHC
- Wall heat flux lowered by ~30% versus inert wall
- Quenching distance not affected by IFHC



► **Results for an isothermal wall with IFHC: well posed, convergent problem**



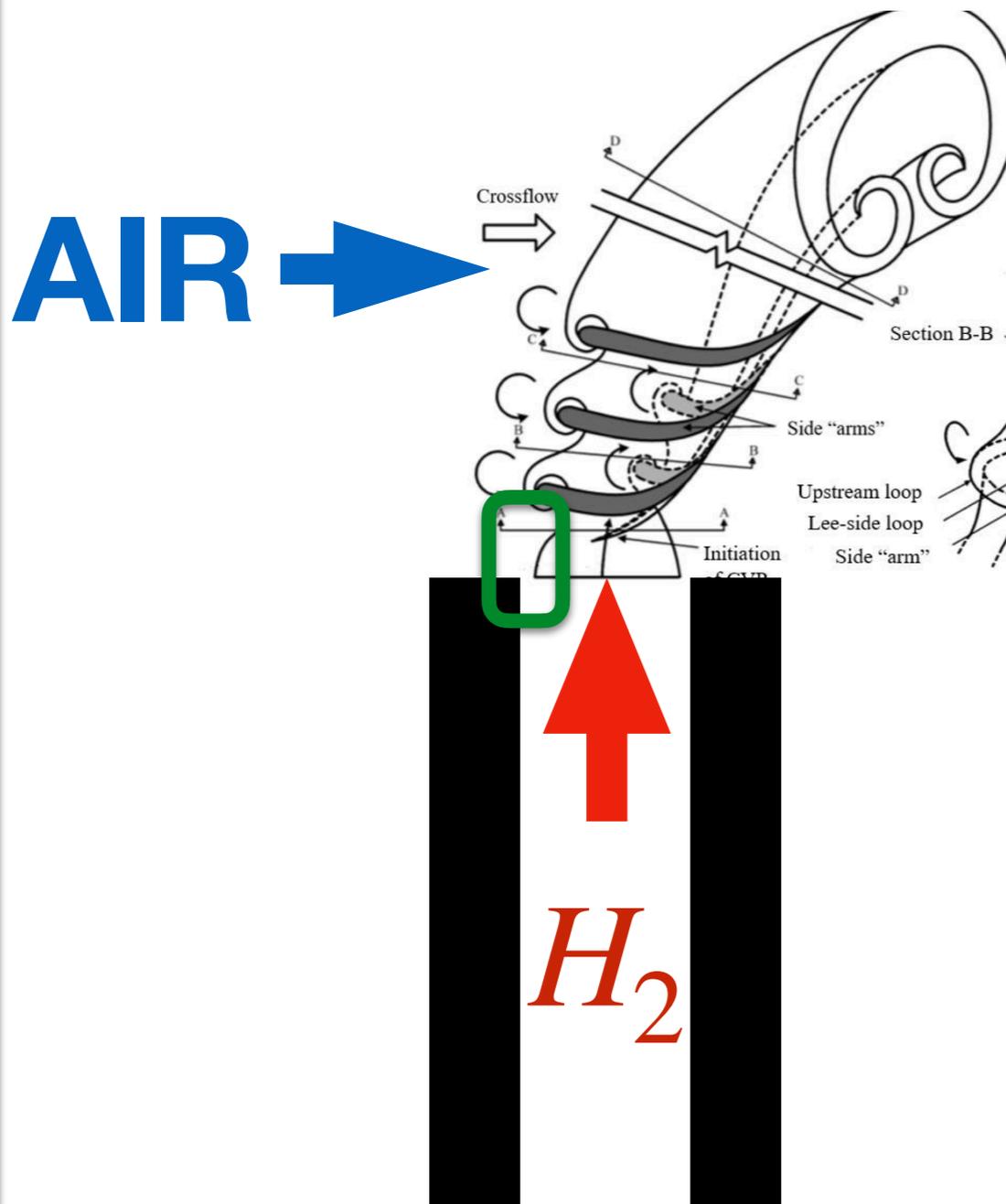
- Mesh influence with the IFHC model
 - 1D HOQ performed for $N_c \in [7; 224]$
- **Grid convergence retrieved**

Yet another CFD issue: flame stabilization

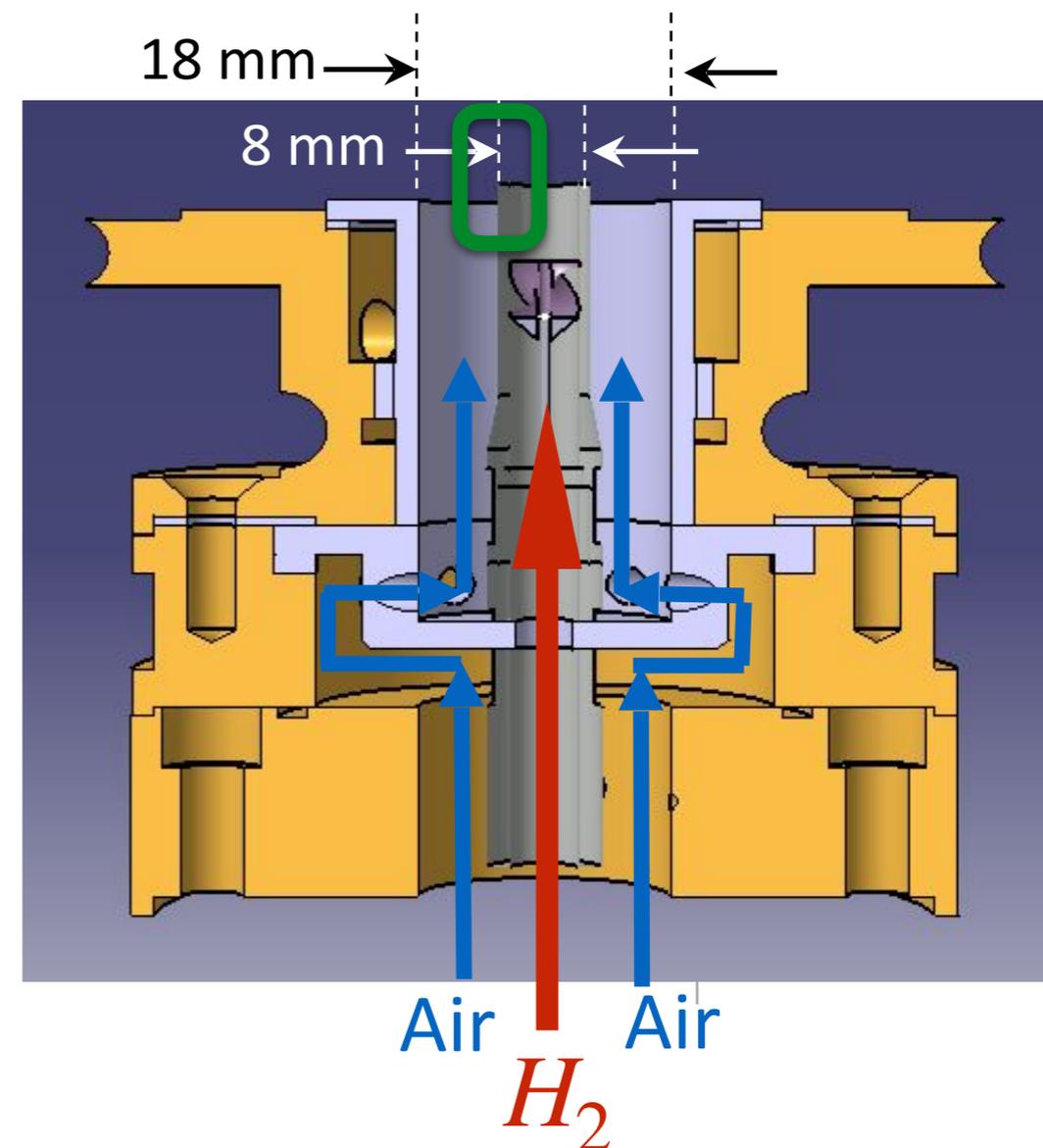
Why worry about stabilization?

- ❖ All H₂ air concepts rely on 'some creative method' to mix H₂ and air fast

MICROMIX

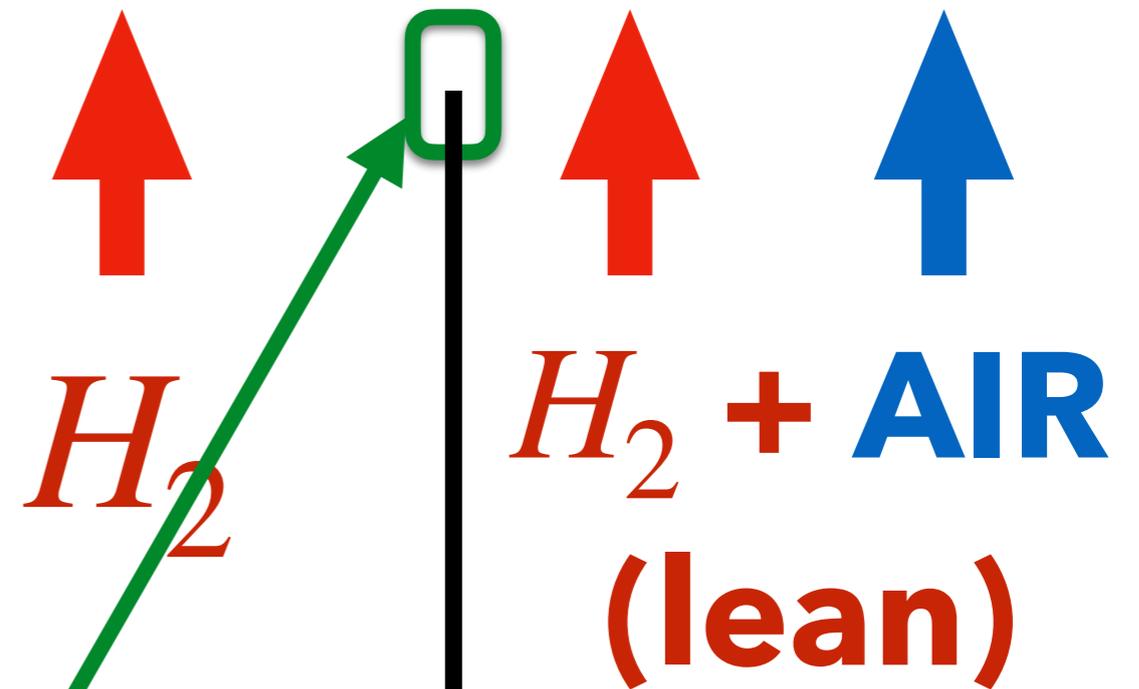
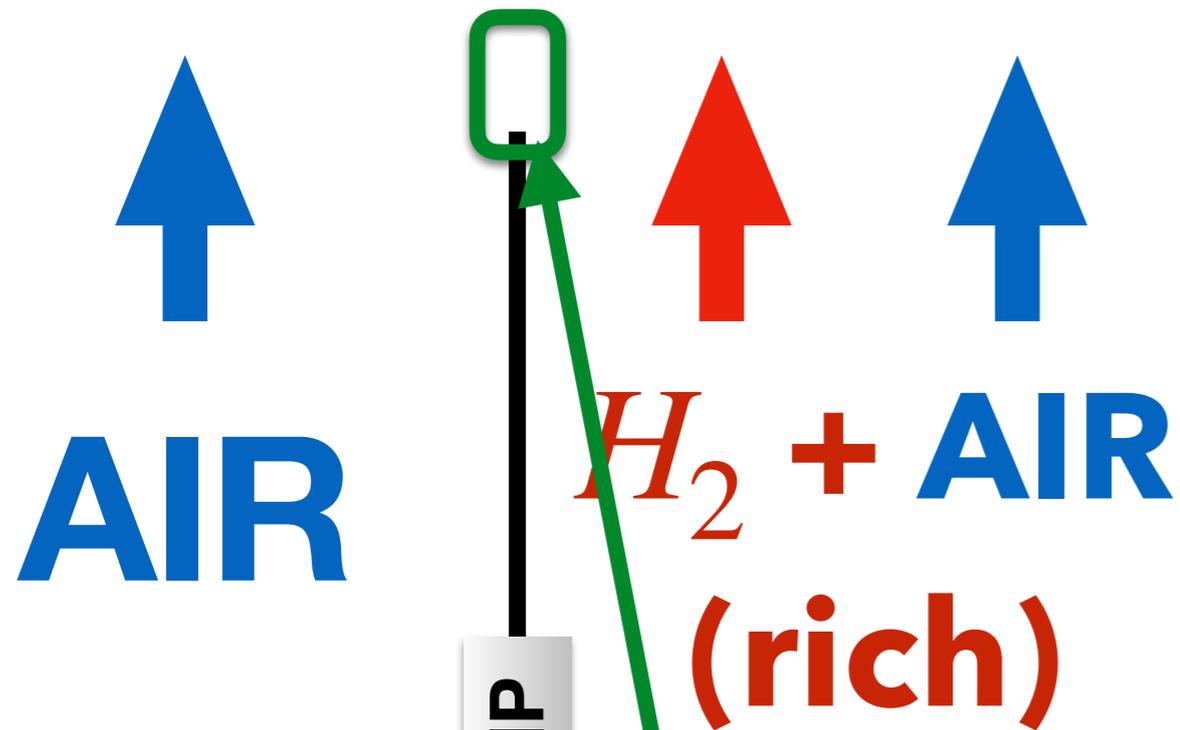


HYLON



SAFRAN TECH

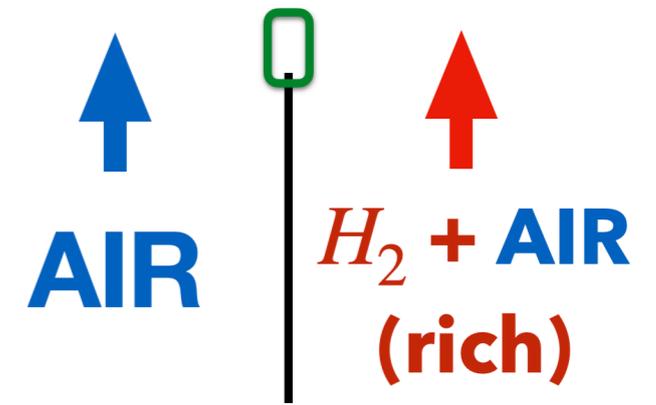
ONERA



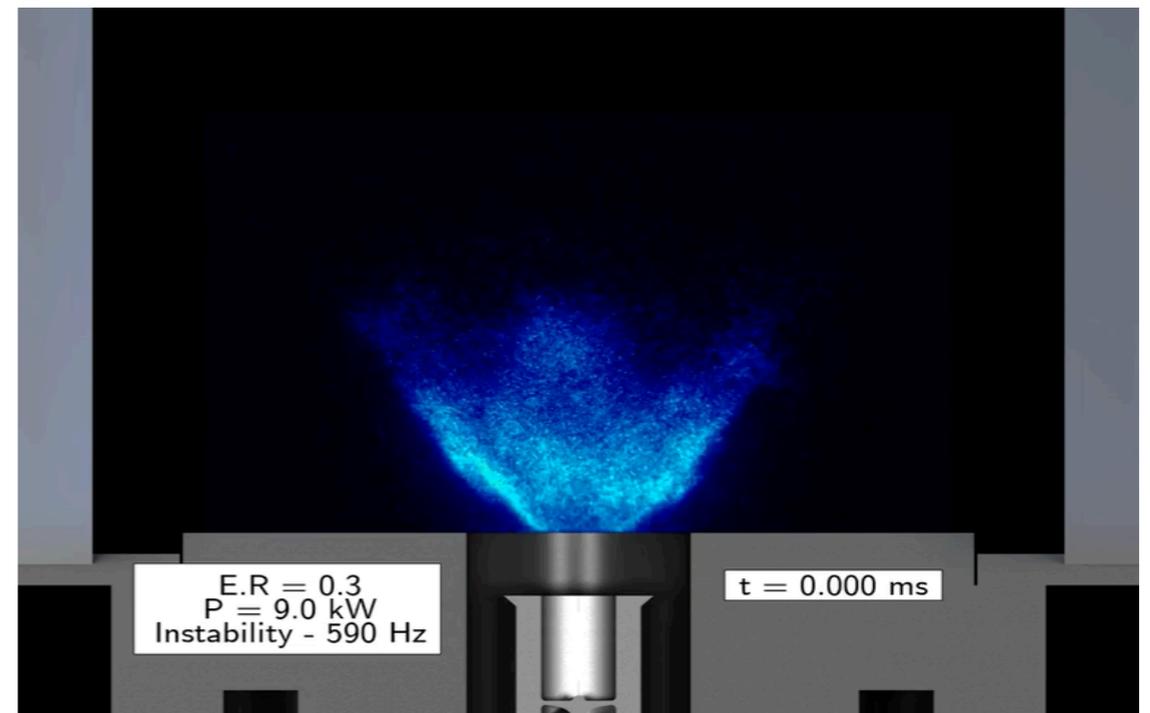
VERY SMALL BUT VERY IMPORTANT
ZONES OF THE COMBUSTOR

Why should we care ?

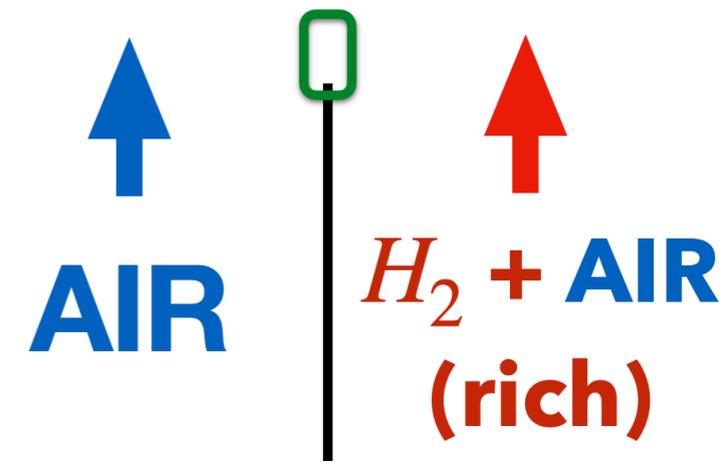
❖ This zone controls flame anchoring:



- ❖ Flame **too close** to the lips: the lips will burn. If they do not burn, we'll have a diffusion flame and too much NO_x. After a while, we may even have flashback
- ❖ Flame **too far** from the lips: the flame might blow off or become unstable -> thermoacoustics

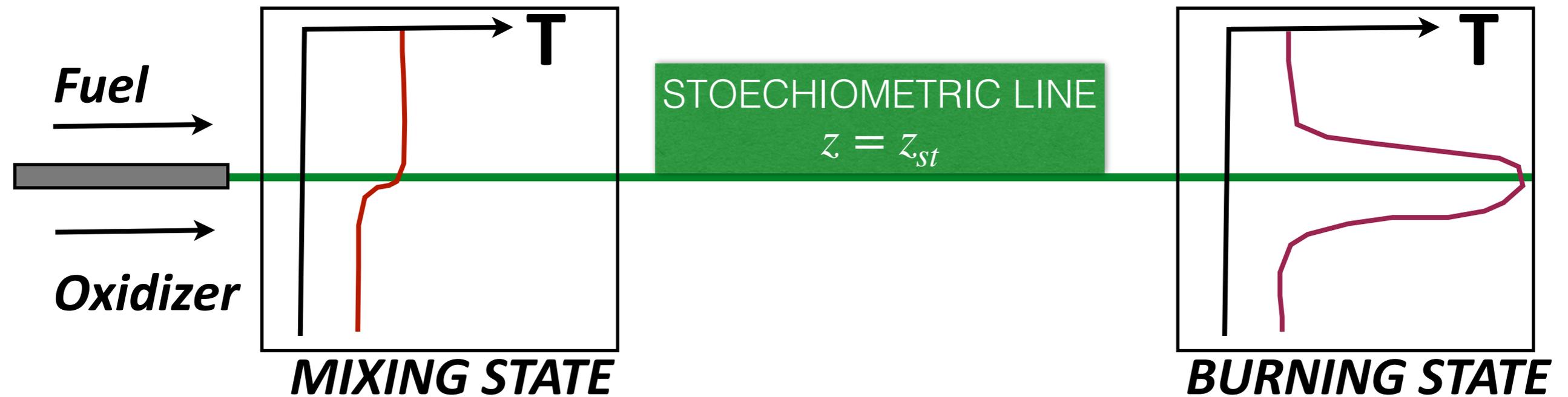


Why is this zone a problem?



- ❖ This zone contains flame elements which are beyond our LES models: 'edge' flames (among which, the 'triple' flames)
- ❖ Edge flames are not included in any turbulent combustion model
- ❖ Edge flames are very small...

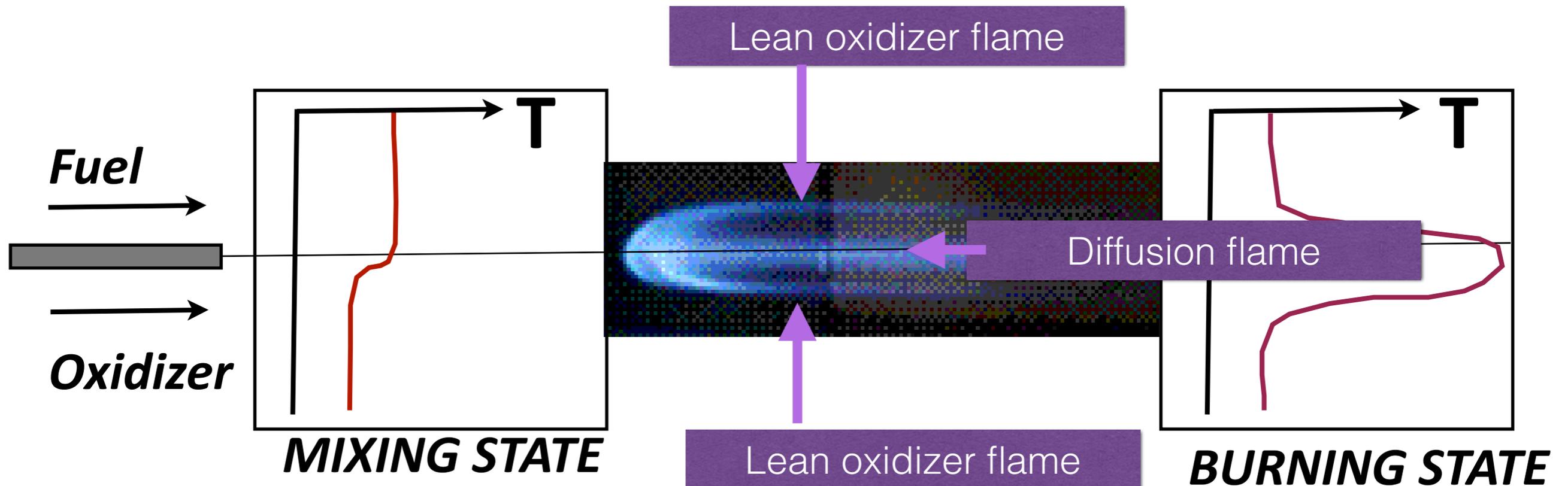
THE TRIPLE FLAME: THE STRUCTURE WHICH SEPARATES IGNITED FROM NON IGNITED DIFFUSION LAYERS



In any system, where you inject pure fuel and pure oxidizer, you start by mixing them with no flame. Later downstream, you burn in a diffusion mode on the stoichiometric line $z = z_{st}$. How ?

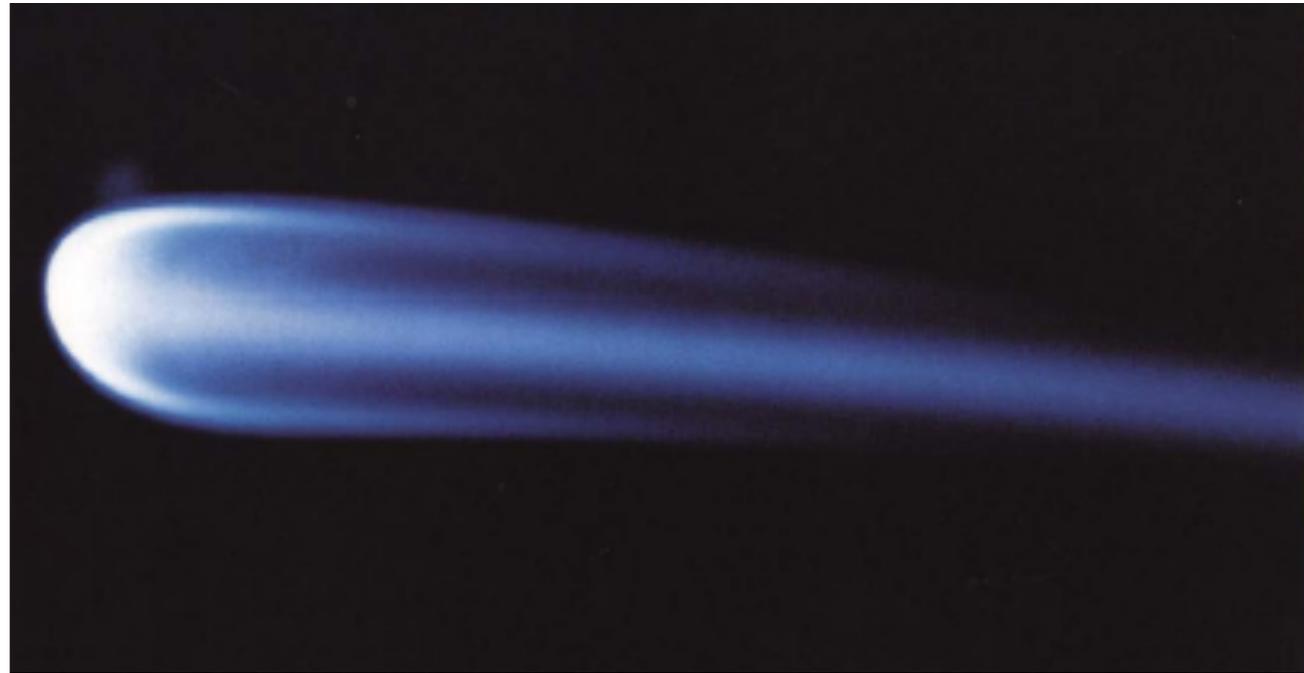
In other words: how does a diffusion flame begin ?

THE CONSENSUS TODAY IS THAT THIS TRANSITION IS PRODUCED BY **TRIPLE FLAMES**:



P. N. Kioni, B. Rogg, K. N. C. Bray, and A. Liñán, "Flame spread in laminar mixing layers: The triple flame," *Combust. Flame* 95, 276, 1993.

- 1/ TRIPLE FLAMES PROPAGATE, HAVE A SPEED
- 2/ TRIPLE FLAMES PROPAGATE **FASTER** THAN PREMIXED FLAMES



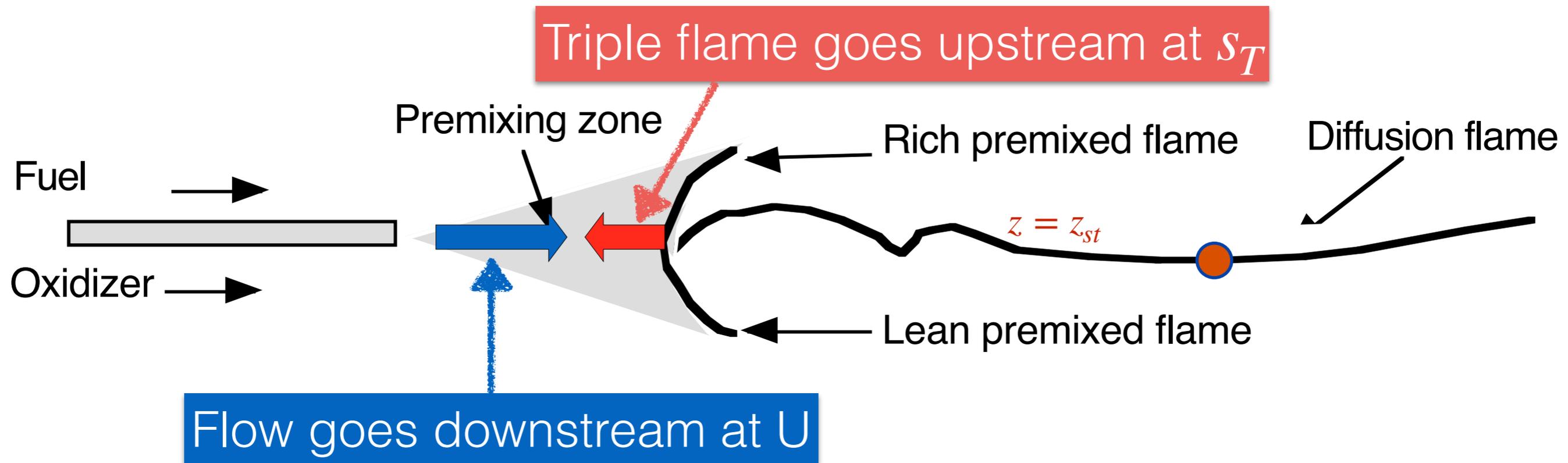
- **A TRIPLE FLAME SPEED SCALES LIKE:**

$$s_{Triple} = s_L^0 \sqrt{\frac{\rho_1}{\rho_2}}$$

Stoichiometric
laminar flame speed

Density ratio

MUNIZ AND MUNGAL: LINK THE STABILIZATION OF DIFFUSION FLAMES TO THE SPEED OF TRIPLE FLAMES



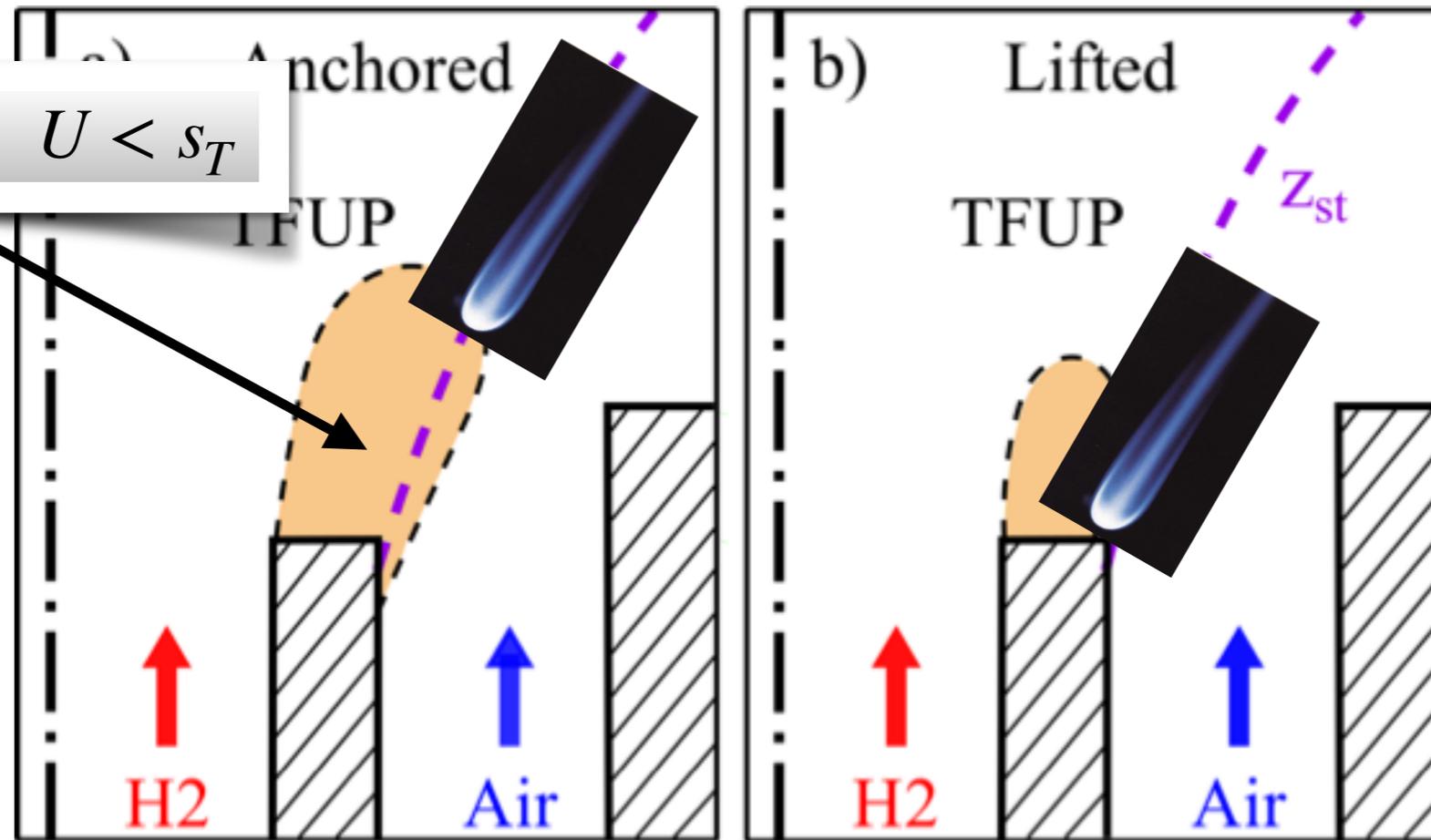
If $U > s_T$ the flame is convected away
If $U < s_T$ the flame goes upstream and attaches to the lips

Remember that the flame MUST travel along the stoichiometric line $z = z_{st}$

L. Muniz and M. G. Mungal. Instantaneous flame-stabilization velocities in lifted-jet diffusion flames. Comb. Flame, 1997, 111, 1-2, 16-31

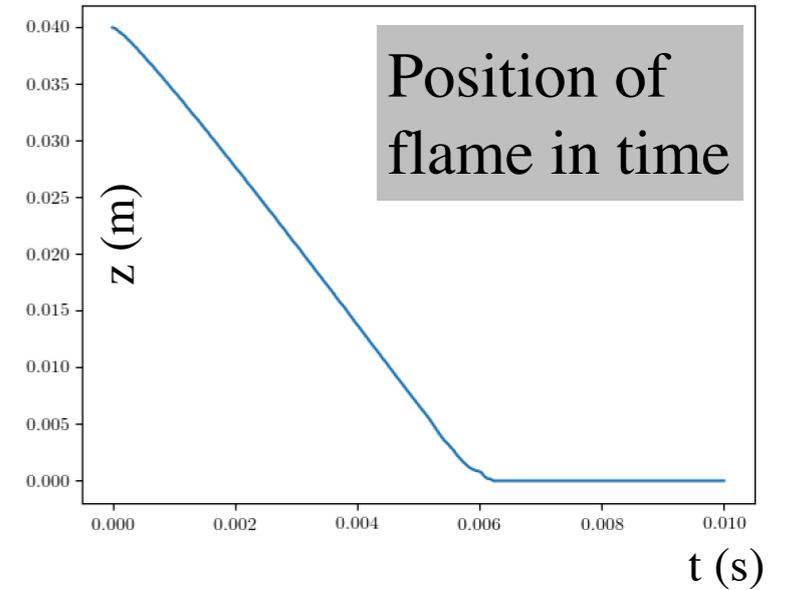
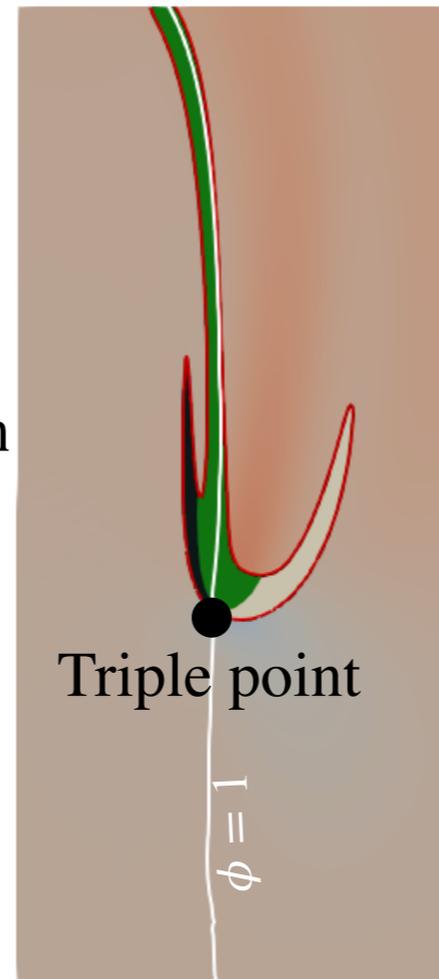
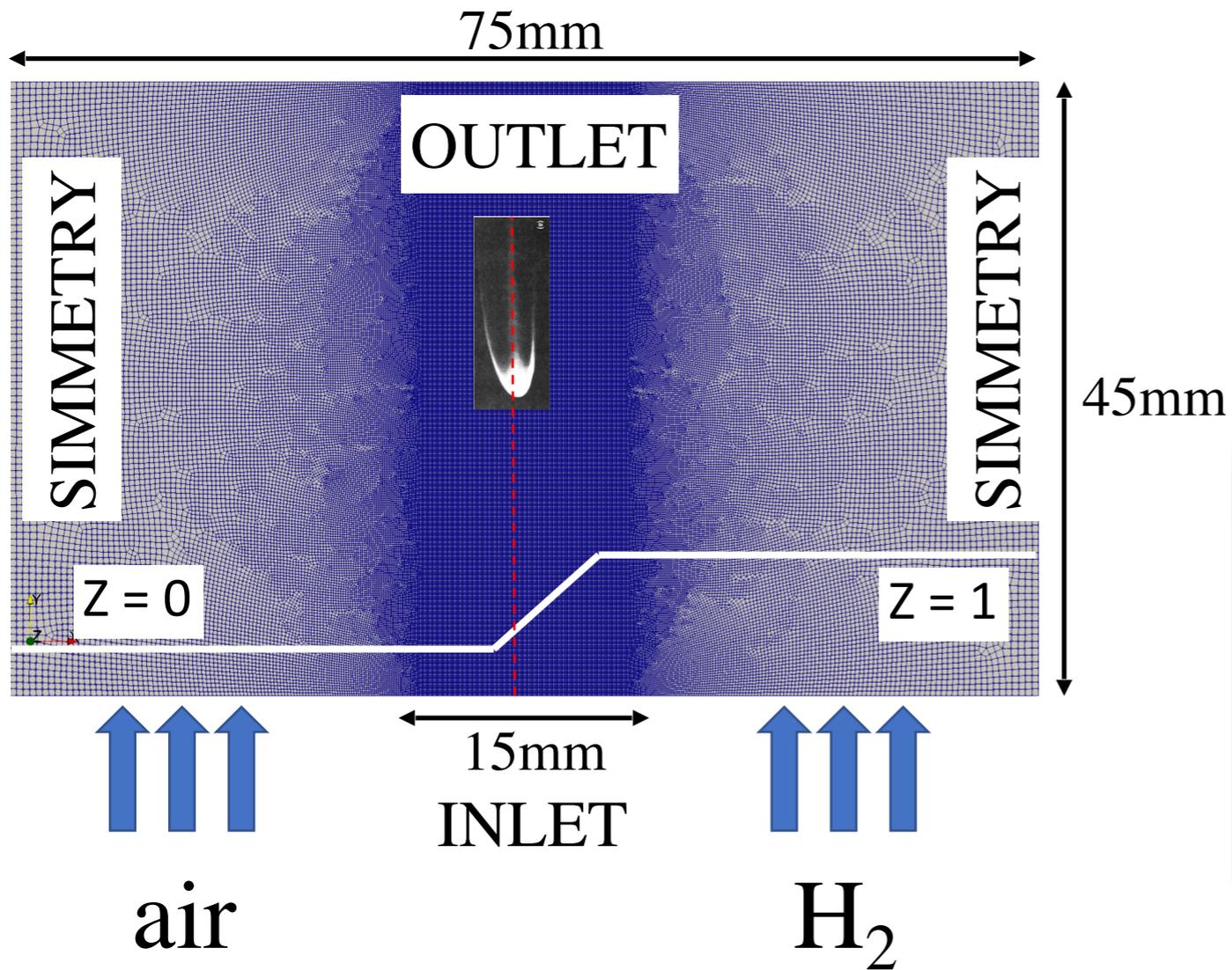
Triple Flame Upstream Propagation (TFUP): for H₂ air flames at 1 bar, the triple flame speed $s_T = s_L^0 \sqrt{\rho_1 / \rho_2}$ is 7 m/s

In this TFUP zone, $U < s_T$



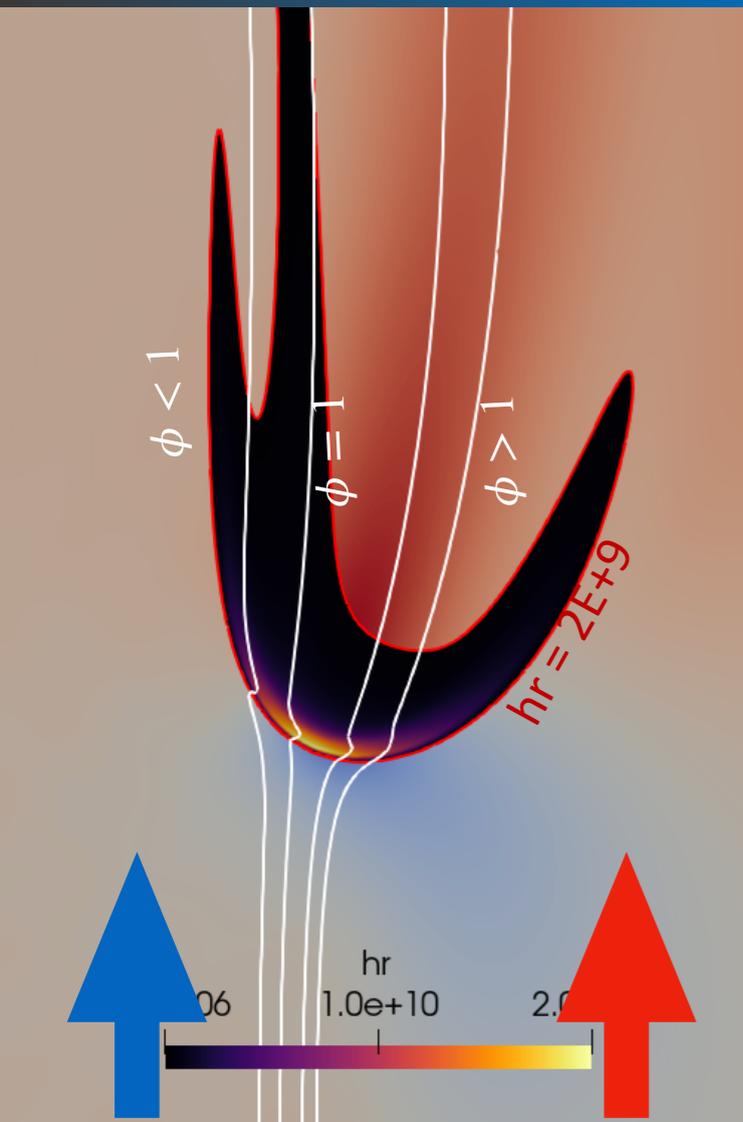
S. Marragou, H. Magnes, A. Aniello, L. Selle, T. Poinso, T. Schuller. Experimental analysis and theoretical lift-off criterion for H₂/air flames stabilized on a dual swirl injector, *Proc. Comb. Inst.*, 2022. 10.1016/j.proci.2022.07.255

DNS of triple H₂ air flames (Aniello PhD)



Flame velocity depending on:

- ∇Z
- Fresh gas temperature
- Pressure



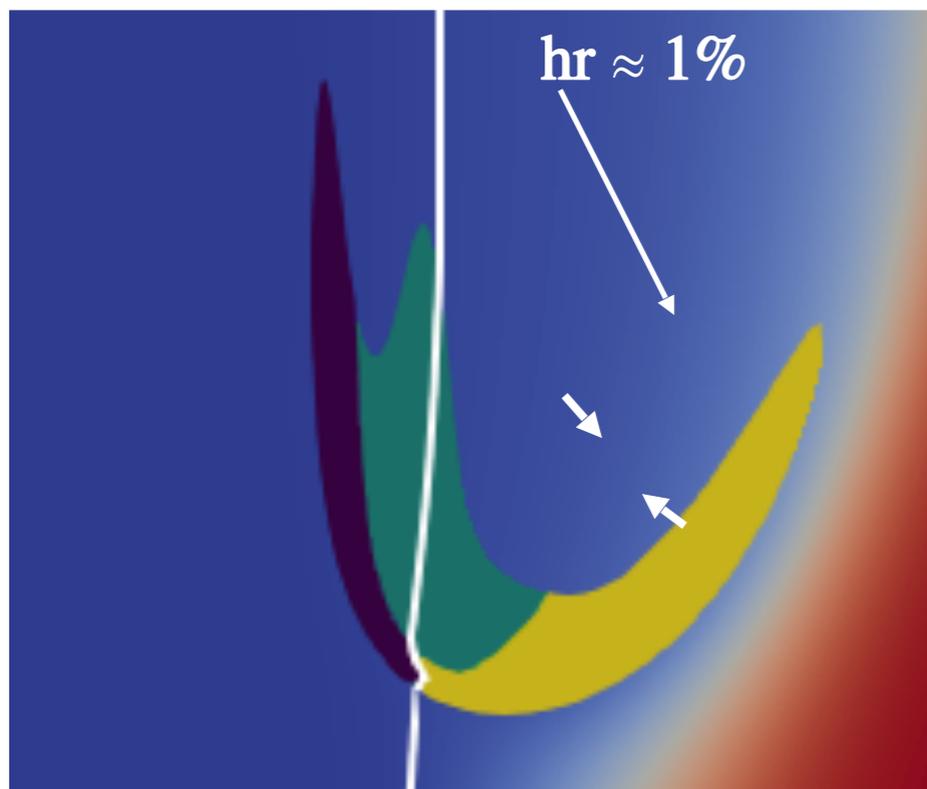
- ❖ We know that these flames exist and control flame stabilization. They are a bit special for H₂
- ❖ Experiments confirm that they provide the right scalings for lift off
- ❖ Can not capture them with LES models: most LES models do not even know what a triple flame is
- ❖ Can capture them with DNS but...

AIR

H₂

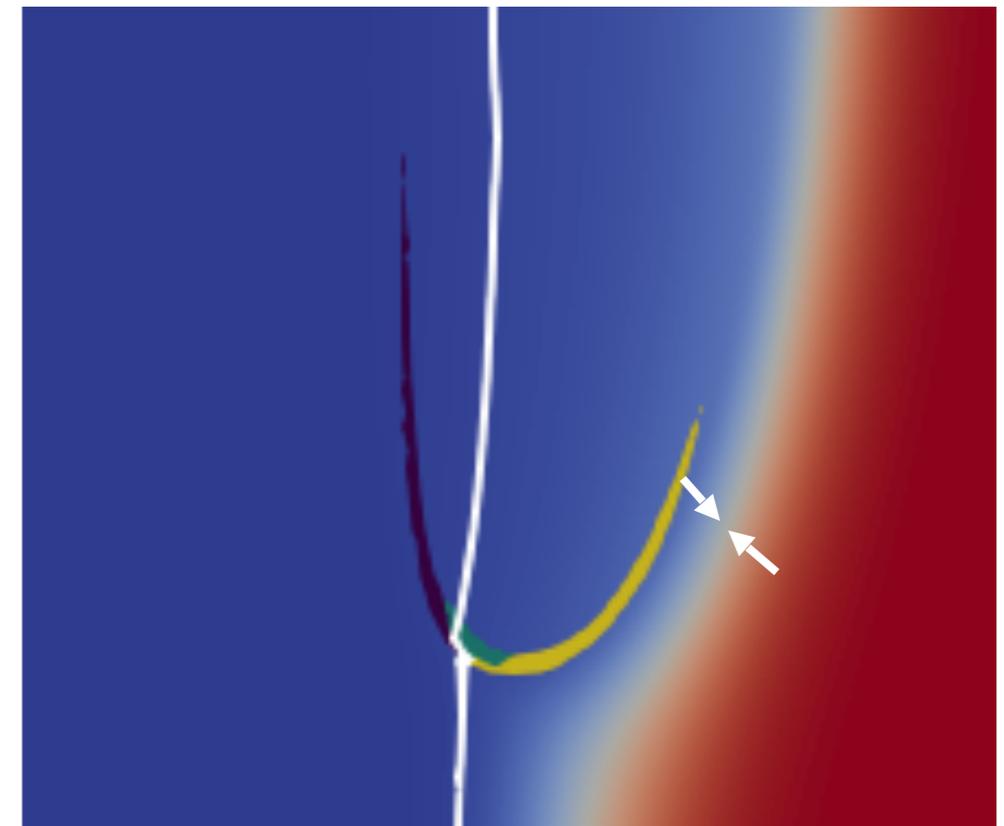
But DNS cant go to high pressures even in 2D...

$\Delta x = 25 \mu\text{m}$



$T = 300 \text{ K} , P = 1 \text{ bar}$

$\Delta x = 8 \mu\text{m}$



$T = 300 \text{ K} , P = 5 \text{ bar}$

'Other' edge flames are expected



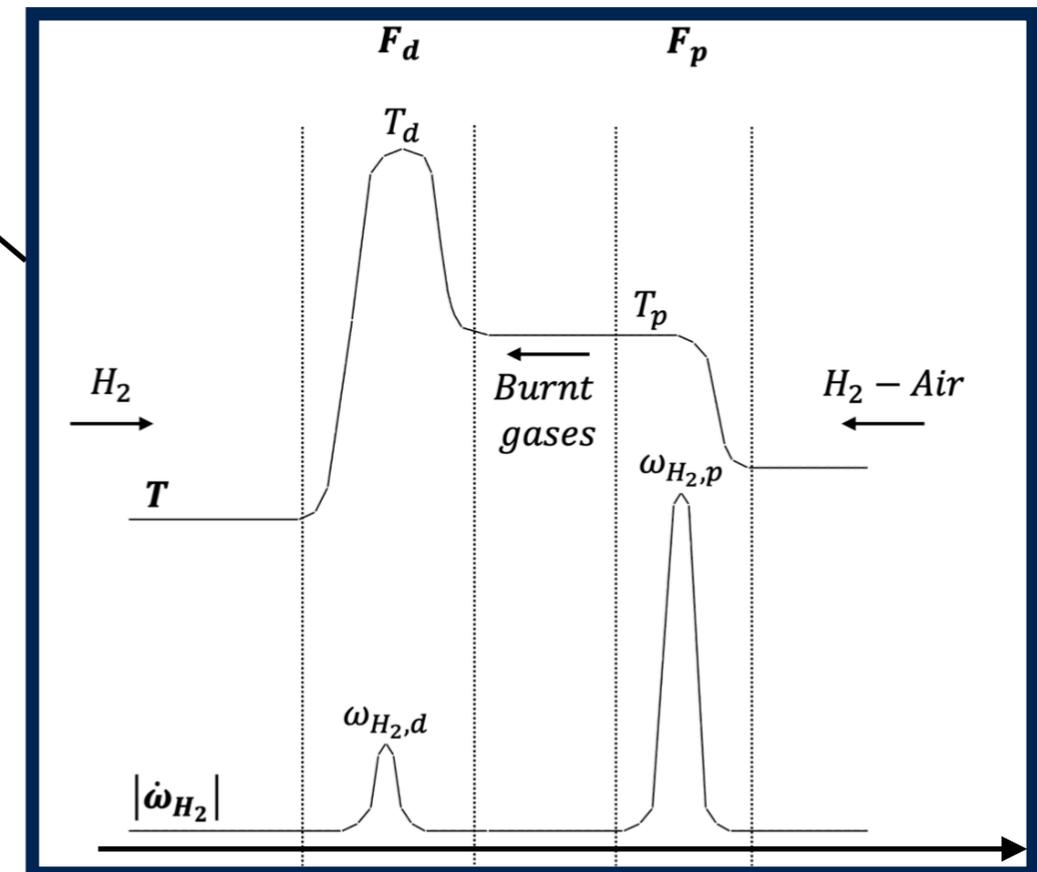
Hot vitiated AIR

Diffusion H_2 /
VITIATED AIR flame

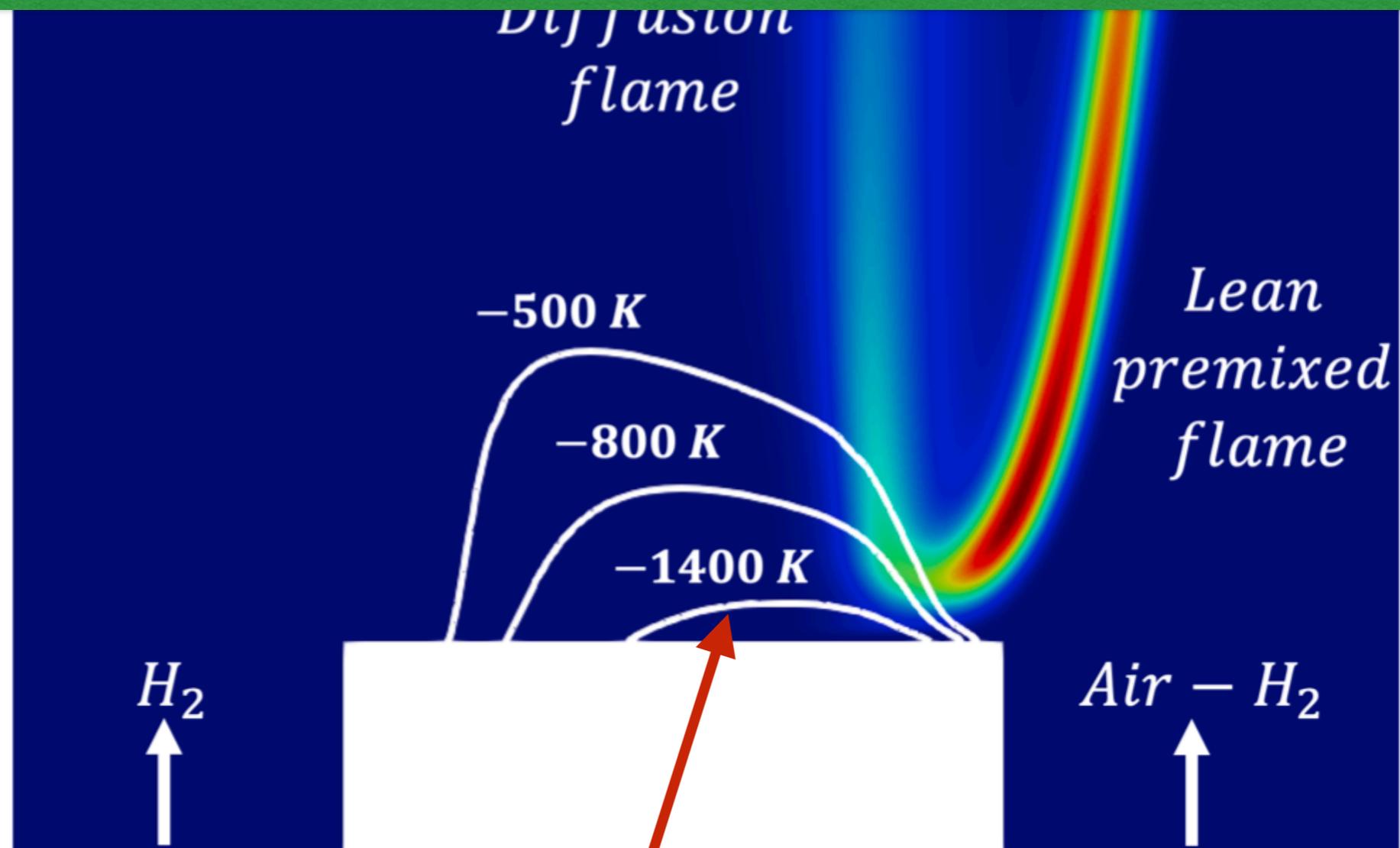
Lean H_2 / AIR flame


 H_2


 $H_2 + \text{AIR}$
(lean)



THESE EDGE FLAMES DO NOT PROPAGATE ANY MORE.
THEY ARE COOLED BY THE LIP AND LOSE A LOT OF
HEAT TO IT -> WE PERFORM CHT COMPUTATIONS



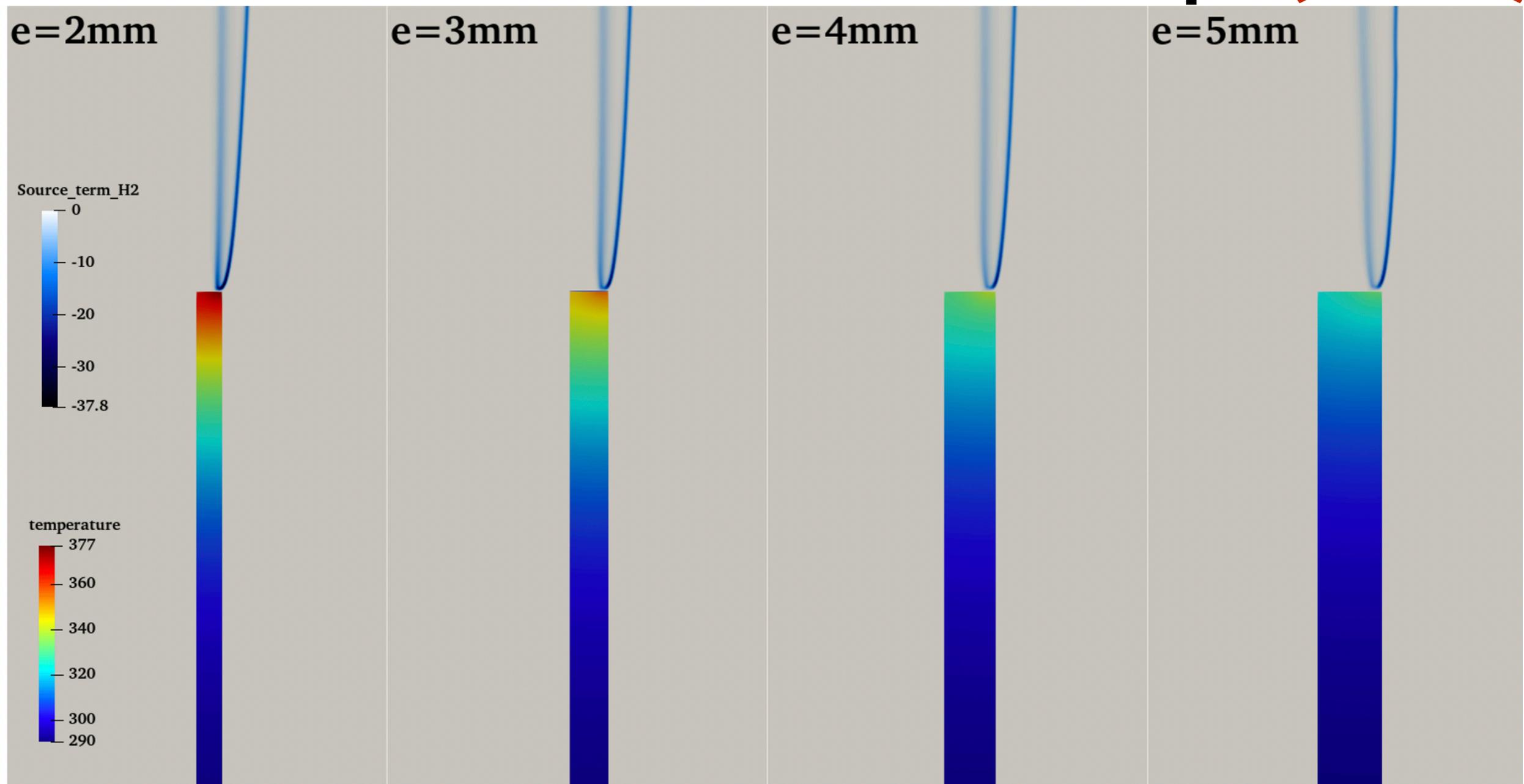
T-Tadia (heat loss)

FULL CHT OF 'RIM-STABILIZED EDGE FLAMES:

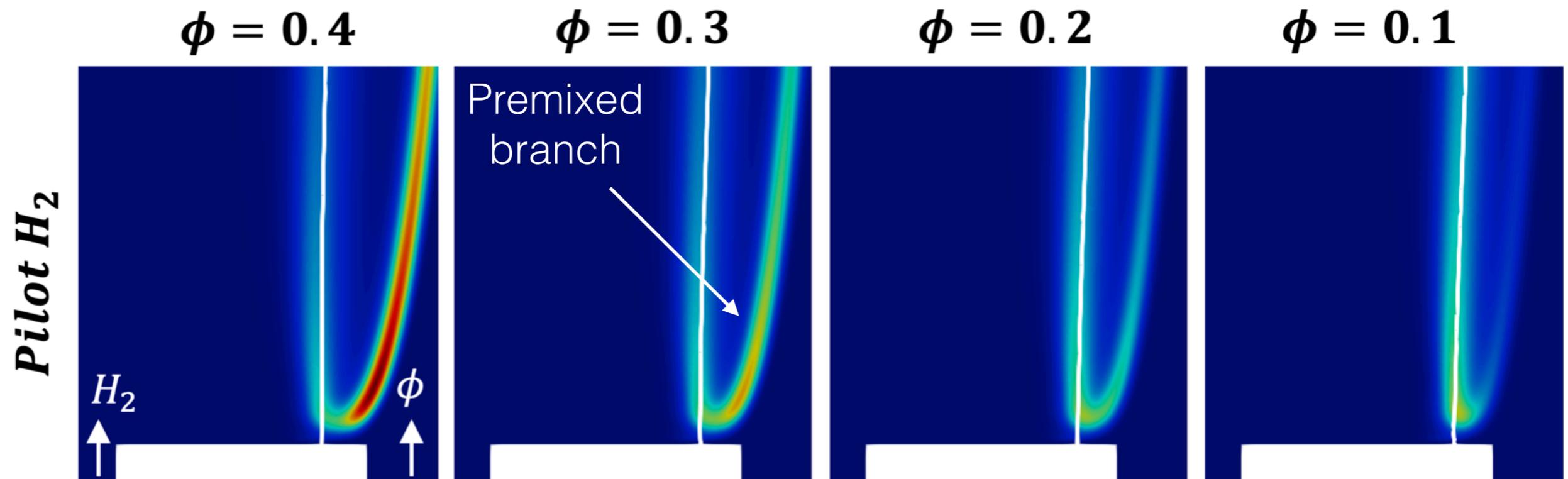



 H_2

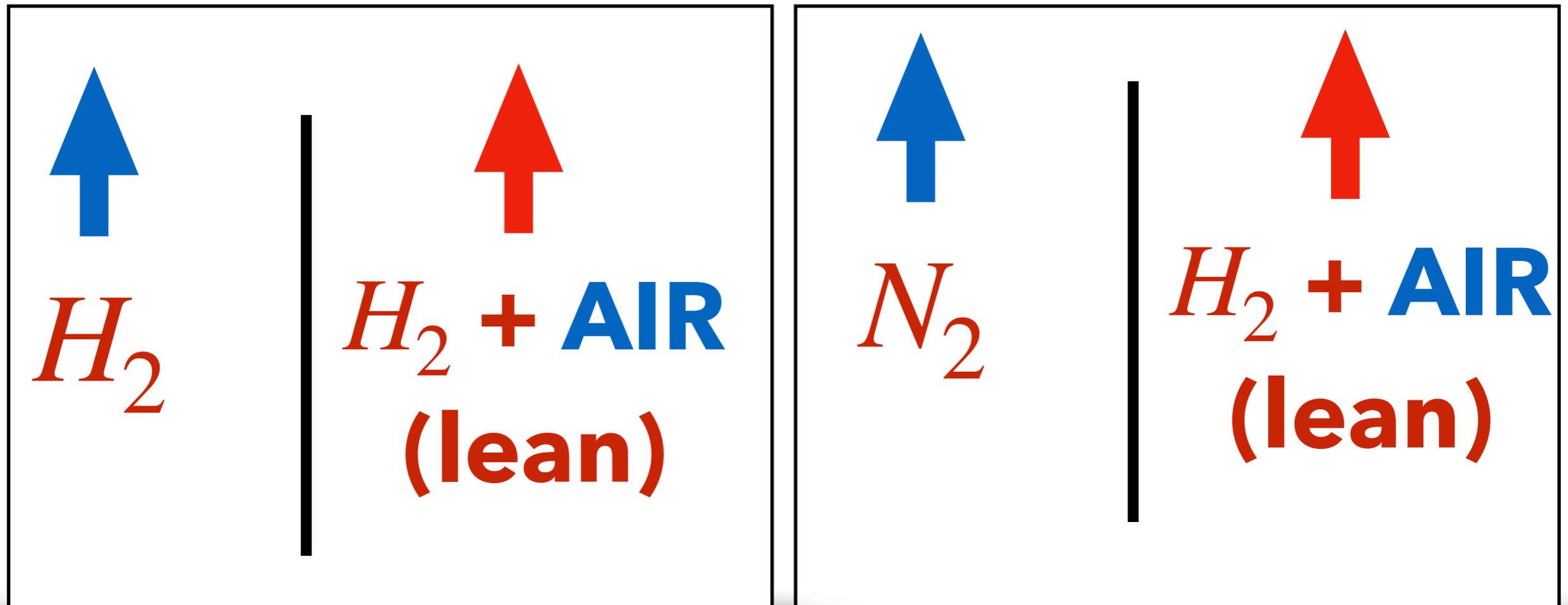
 
 $H_2 + \text{AIR}$
(lean)



THE STRUCTURE OF THESE RIM STABILIZED EDGE FLAMES DEPENDS ON THE EQUIVALENCE RATIO OF THE PREMIXED BRANCH



AND THE PURE INJECTION OF H₂ HELPS TO STABILIZE
THE PREMIXED BRANCH: COMPARING H₂ WITH N₂
INJECTION IN THE LEFT STREAM



GOING TO N₂, THE PREMIXED BRANCH QUENCHES BELOW $\phi=0.25$, CAUSING LEAN BLOW OFF (LBO)

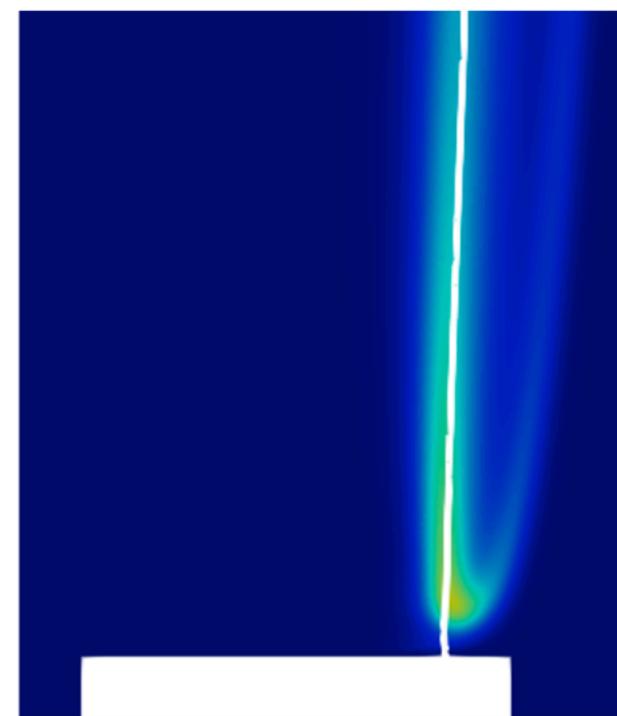
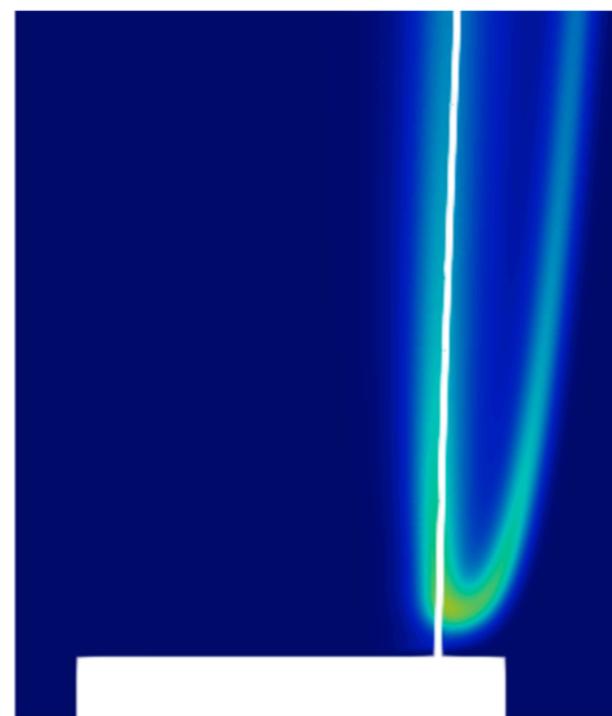
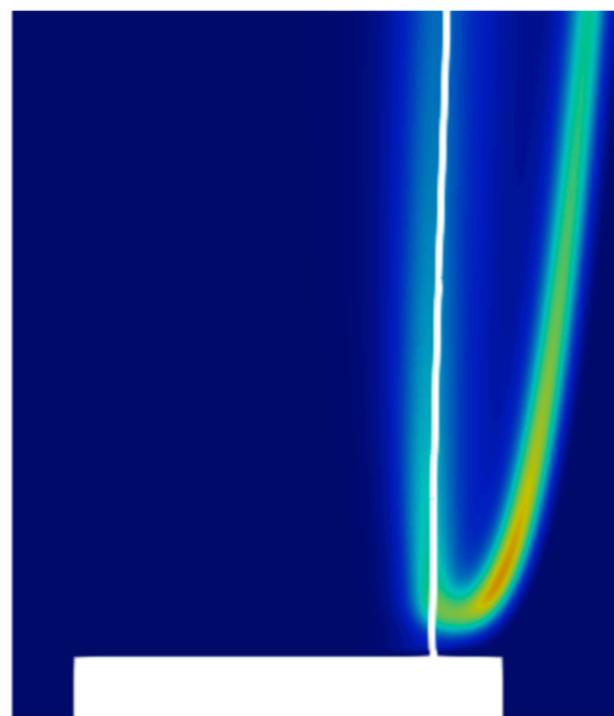
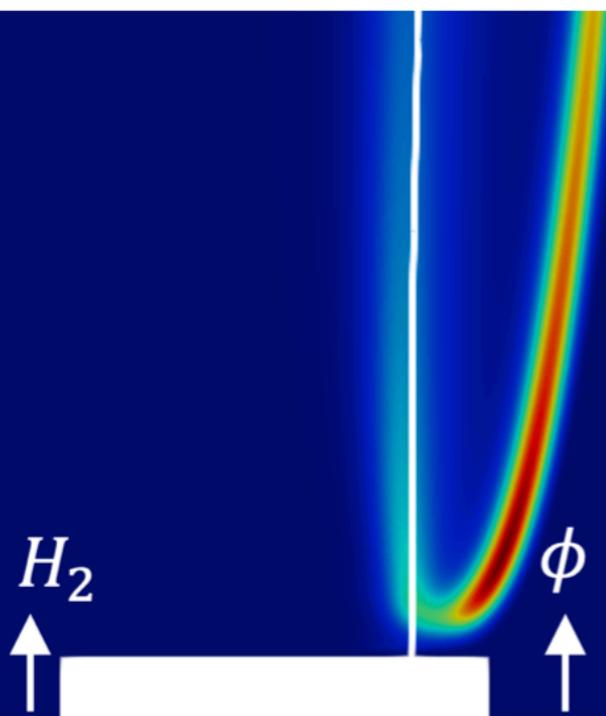
$\phi = 0.4$

$\phi = 0.3$

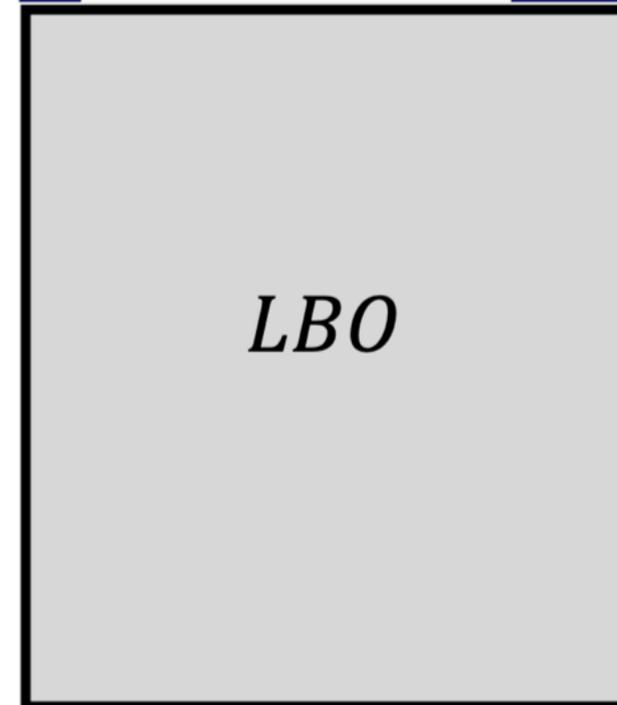
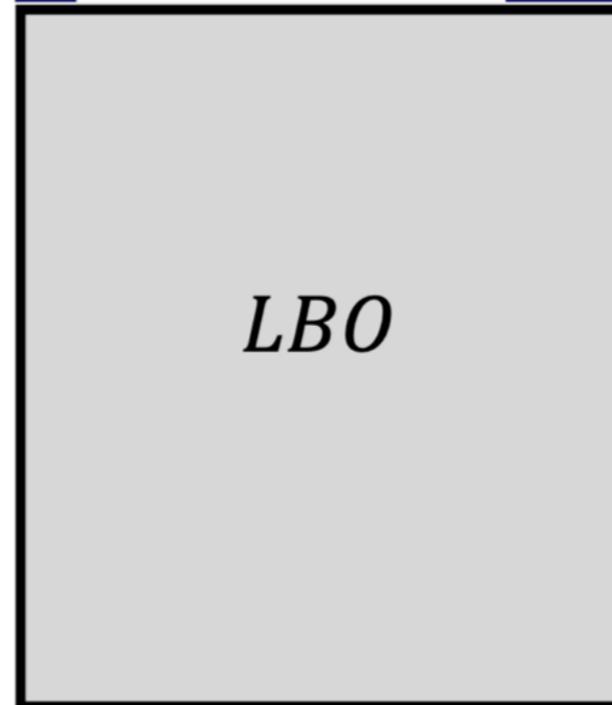
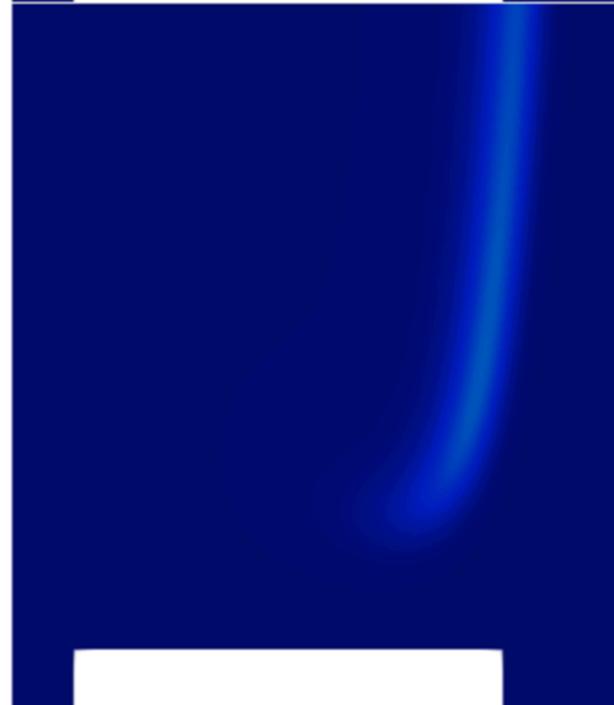
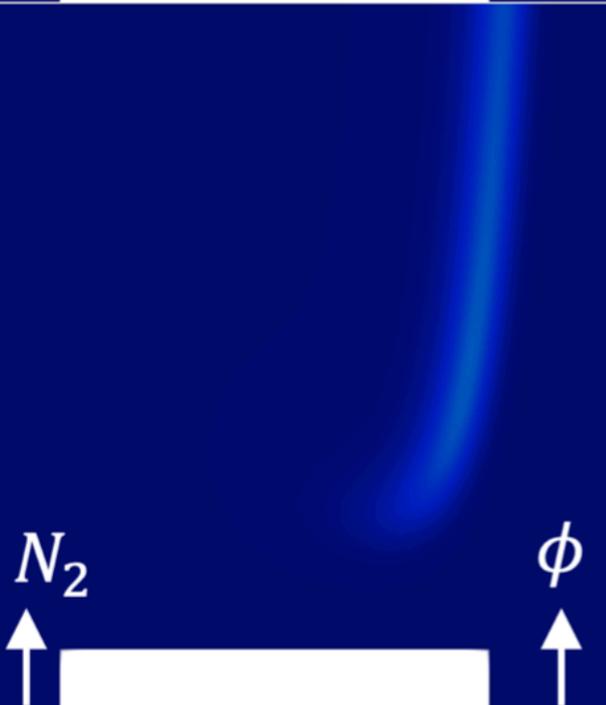
$\phi = 0.2$

$\phi = 0.1$

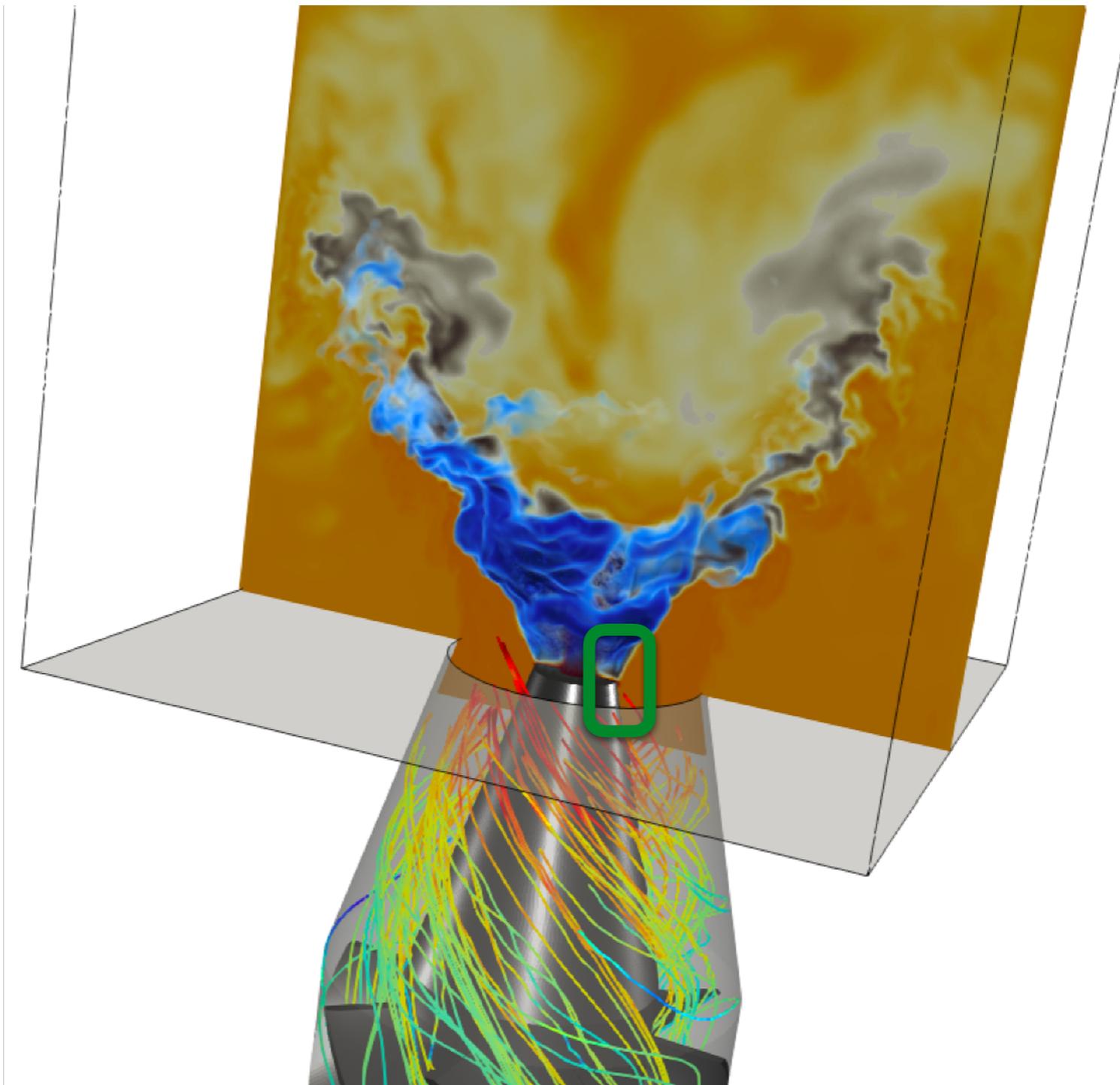
Pilot H₂

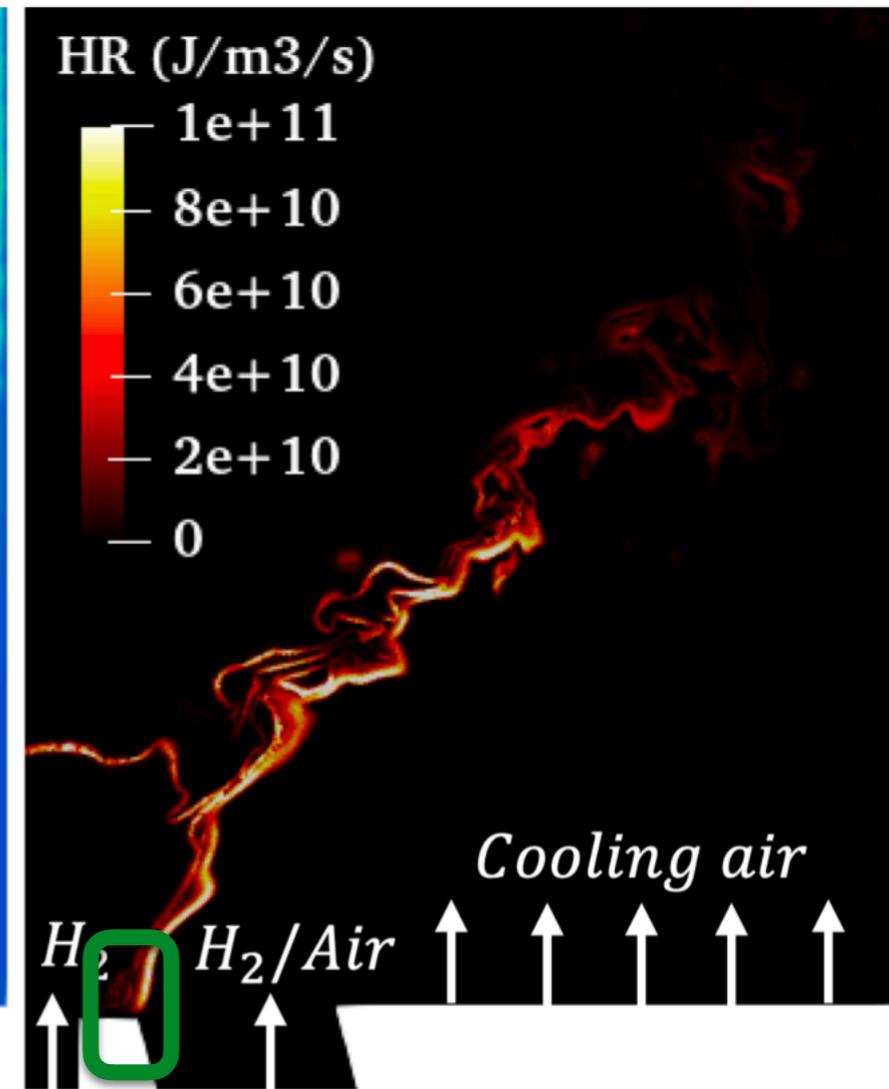
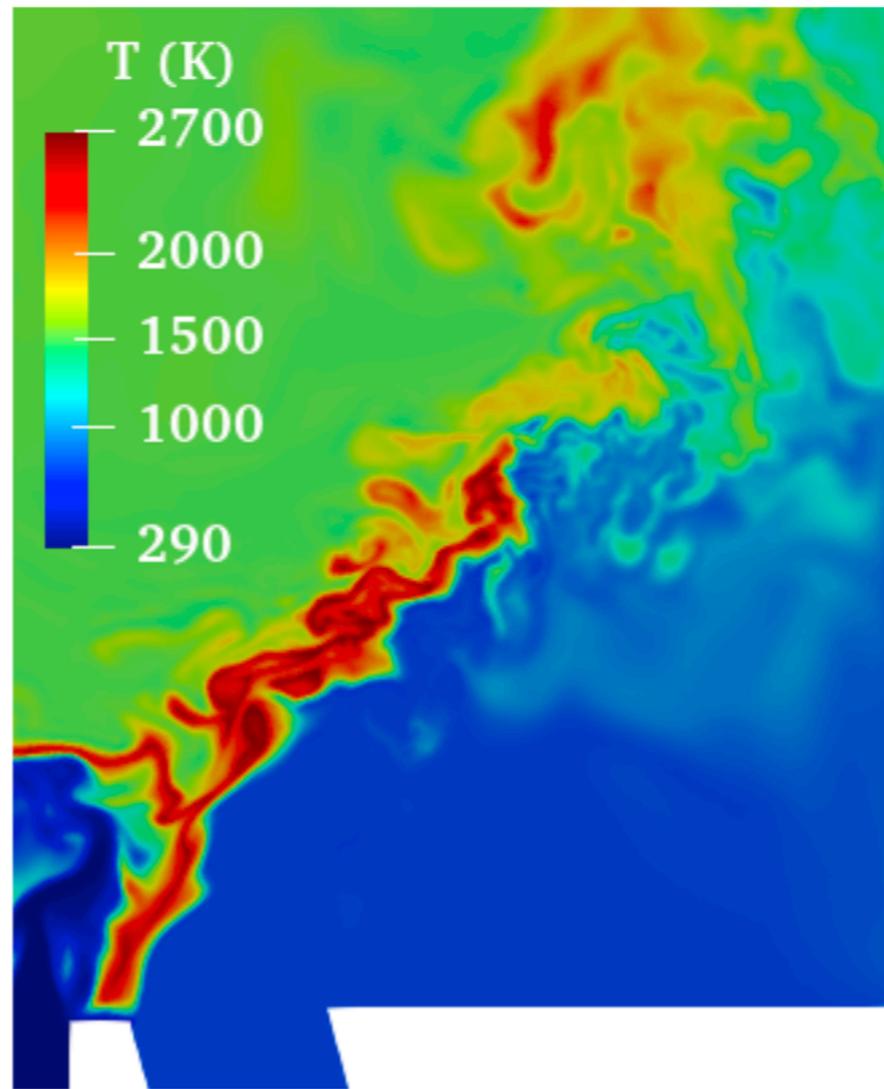
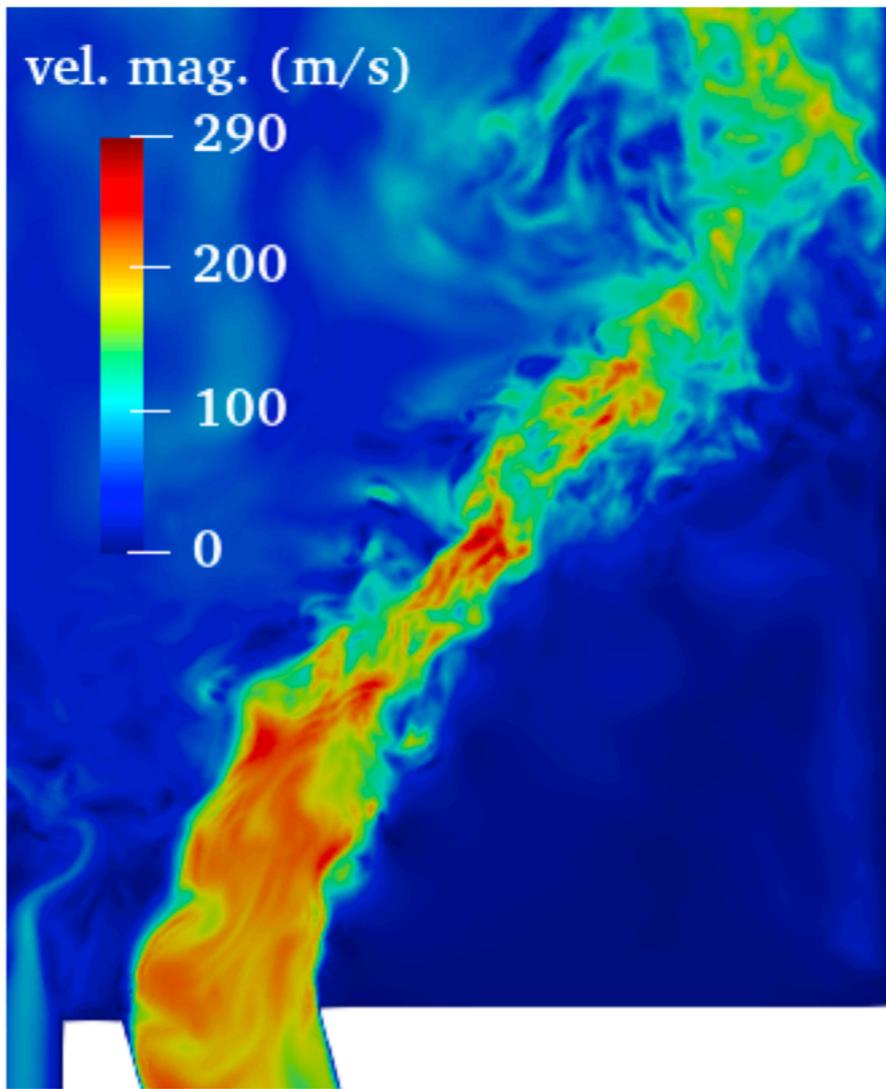


Pilot N₂



Does LES see these flames ?





CONCLUSIONS

- We must re-design most combustion chambers for H₂ flames. Simulations will be essential
- This will not be so easy as H₂ combustion gathers all exotic concepts in combustion theory: multi-regime flames, thermodiffusive instabilities, detonations, edge flames, ultra fast flames + (of course) turbulence...

