

# **Optimizing Aerodynamic Design Problems**

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Joint work with

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### **Remembering Gene Golub Around the World 2008**



## In memory of Gene Golub (February 29, 1932 - November 16, 2007)





## Problem description

Bound-constrained direct minimization

Description of the methods

Dumerical Experiments in CUTEr (academic testing env.)

Dumerical Experiments in OPTaliA (industrial optimization env.)

Conclusions





(NLP) minimize f(x)subject to  $x \in X \cap \{x \in \mathbb{R}^n : c(x) \le 0\}$ 

- X is a region where f(x) and c(x) can be evaluated by the underlying CFD simulation
- ▶ f, c are expensive black boxes mins, hours, days, weeks
- nonlinear and nonconvex
- Gradient is available (obtained by Adjoint state system)
- Hessian has to be approximated by the optimizer if necessary





- Application: optimize design of a wing shape (collaboration with Airbus)
- Goal: find the **best strategy** to solve these problems
  - Improve existing strategies
    - Direct minimizer DOT
      - BFGS method, CG method
    - Surrogates approach
      - Kriging and Co-Kriging Model Framework
  - Generalize to constraints
- First step: compare direct solver DOT with a set of well-known optimization codes free for academic use





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- Direct minimization:
  - approach to minimize the objective function by evaluating directly the true problem function and/or gradient values at each step
- Comparison: DOT with a set of well-known optimizers (L-BFGS-B, TN-BC, Lancelot, IPOPT, DONLP2)
- Important points for fair comparison:
  - Use the same stopping criteria
  - Parameter choice: usage of default values
  - Use the same information of the function (funct. value, first derivatives)





- Used stopping criteria:
  - Problem solved successfully
    - Infinity norm of projected gradient <= 10e-5</p>
  - Problem not successfully solved
    - Termination by solver:
      - Found no solution
      - Stuck with projected gradient > demanded accuracy
    - Termination by user:
      - Number of iterations > 100000 / 200
      - CPU-Time > 1800 s / 24 h





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- Selection of the solvers (source code available, no Hessian necessary)
  - DOT BFGS Method, Fletcher Reeves Conjugate Gradient Method (Vanderplaats Research & Development, Inc.)
  - L-BFGS-B Limited Memory BFGS Method (Richard H. Byrd, Peihuang Lu, Jorge Nocedal, Ciyou Zhu)
  - TN-BC Truncated Newton Method (Stephen G. Nash)
  - Lancelot B Trust Region method SR1, BFGS, PSB update (Nicholas I. M. Gould, Andrew Conn, Philippe L. Toint) --- <u>at a price for commercial use</u>
  - IPOPT Interior Point Method (Andreas Waechter)
  - DONLP2 SQP Method (Peter Spellucci)



## **Description of the methods**



• Algorithmic components of the solvers

	Framework		Linear algebra		needs
	Line Search	Trust Region	direct	iterative	Hessian approximation
DOT BFGS	Х		Х		yes
DOT FR	Х			х	no
L-BFGS-B	Х		Х		yes
TN-BC	Х			х	yes
Lancelot B		х		х	yes
IPOPT	Х		Х		yes
DONLP2	х		Х		yes





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- CUTEr: (a <u>Constrained and Unconstrained Testing Environment revisited</u>)
  - Testing environment to compare optimization and linear algebra solvers
  - Contains a large collection of test problems in SIF (Standard Input Format)
  - Provides ready-to-use interfaces to existing solvers (algorithms are not included, have to be implemented)
  - Possible to create new interfaces
- Overview of used test problems
  - 76 out of 128 bound constrained problems provided by CUTEr
  - Nbr. of variables: 3 to 15625
  - Type of objective function: quadratic (32), sum of squares (19), other (25)





• Results in terms of function + gradient evaluations

	Nbr. of won test cases		
DOT BFGS	2		
DOT FR	1		
L-BFGS-B	9		
TN-BC	1		
Lancelot B SR1	49		
Lancelot B BFGS	38		
Lancelot B PSB	48		
IPOPT	0		





### **Numerical Experiments in CUTEr**

• Results in terms of function + gradient evaluations







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## All steps take place in the same environment (virtually)









• Convergence history (each func evaluation) of true Airbus function (n=6)







#### **Numerical Experiments in OPTaliA**

• Function and gradient of a one-dimensional Airbus function







### **Numerical Experiments in OPTaliA**

• Function and gradient of a one-dimensional Airbus function







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- Behaviour of solvers on true aerodynamic functions different from academic test cases
- Many local minima
- Noise detected in function and gradient, due to
  Single precision in CFD simulation
  - Truncation of simulation process
- Gradient can be inexact, depending on objective function and parameters in CFD simulation (angle of attack, mach number, coefficients in the adjoint state system)
- Use of surrogates necessary to solve these kind of problems





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- Idea: replace expensive function by a surrogate / model which is (well enough) representing the function
- Advantages: surrogates much cheaper, continuous, differentiable
- Enables the use of sophisticated optimization routines and possibly of more global optimization techniques
- Two general types: Functional models (generated by fitting into sampled data), Physical models (based on simplification of the particular physical system)





## **General Surrogate Optimization Framework**





- Compare / improve state-of-the-art techniques to handle design optimization with surrogates
  - SMF Surrogate Management Framework

[Booker, Dennis, Frank, Serafini, Torczon, and Trosset, A Rigorous Framework for Optimization of Expensive Functions by Surrogates (1999)]

TR – Trust Region Framework

[Conn, Gould, Toint (2000), NASA AIAA papers Alexandrov, Lewis (2000)]





## **Optimization Using Surrogates**

- Surrogates to consider:
  - Response Surface Model

[R.H. Myers, D.C. Montgomery, Response Surface Methodology, second edition, John Wiley & Sons, Inc., 2002]

## Neural network

[Howard Demuth, Mark Beale, Neural Network Toolbox User's Guide, The Mathworks, Inc., 1994]

## Kriging

[Lophaven, S.N., Nielsen, H.B., Sondergaard, J., DACE: a Matlab Kriging toolbox, version 2.0, 2002]

## Radial Basis Function

[Mark A. Abramson, Matlab Toolbox: RBF version 1.0 User's Guide, 2006]



- Surrogates are necessary and very useful in the context of Aerodynamic Design Optimization
- Better surrogates provide better predictions of the real function value and hence fewer expensive function evaluations are needed
- Task:
  - Find suitable surrogate and suitable inner solver which minimizes the surrogate inside a good outer algorithm which minimizes the real function





## Thank you for your attention!

